

Robust Inversion-based Feedforward Control with Hybrid Modeling for Feed Drives

Haijia Xu, Christoph Hinze, Andrea Iannelli, *Member, IEEE*, and Alexander Verl, *Member, IEEE*

Abstract—This paper presents a robust feedforward design approach using hybrid modeling to improve the output tracking performance of feed drives. Geared towards the use for feedforward design, the hybrid model represents the dominant linear dynamics with a flat analytical model, and captures the output nonlinearity by Gaussian process regression. The feedforward control is based on the model inversion, and the design procedure is formulated as a signal-based robust control problem, considering multiple performance objectives of tracking, disturbance rejection and input reduction under uncertainties. In addition, the technique of structured μ synthesis is applied, which allows direct robust tuning of the fixed-structure feedforward gains and ensures the applicability in industrial hardware. The proposed methodological approach covers the entire procedure from modeling to control architecture selection and weights design, delivering an end-to-end strategy that accounts for performance and robustness requirements. Validated on an industrial milling machine with real-time capability, the proposed robust controller reduces the mean absolute tracking error in the transient phase by 83% and 63% compared to the industrial standard baseline feedforward and the nominal design, respectively. Even with a variation of 20% in the model parameters, the robust feedforward still reduces the error by 58% in the worst case with respect to the baseline.

Index Terms—Drive control, hybrid modeling, mixed uncertainty, robust feedforward control.

I. INTRODUCTION

MODERN manufacturing is subject to ever-increasing demands for high productivity and tight part tolerances. Feed drives, which are the main motion-generating components, are required to achieve high-precision tracking of a high-speed motion profile. Industrial control systems are predominantly of the PID type, mostly combined with velocity and acceleration feedforward to improve output tracking behavior [1], [2]. This standard approach is effective for mechanical systems that resemble rigid body dynamics, but has limited performance for flexible machine structures, which are increasingly evident in highly dynamic motion.

Inversion-based feedforward control has been widely investigated to compensate for known higher-order dynamics of

the plant to achieve accurate output tracking. These include the zero-phase-error tracking control (ZPETC) [3], [4], the zero-magnitude-error tracking control (ZMETC) [5]–[7], the parameter varying feedforward [8], [9], and the optimization-based inverse feedforward with neural networks [10], [11]. These approaches require a rather precise dynamics model with significantly increased identification effort. Various adaptive feedforward controllers have been studied to eliminate the influence of model uncertainty and to reduce the commissioning effort by updating the feedforward gains online [12]–[17]. The convergence of the gain adaptation requires that the reference trajectory is sufficiently informative and satisfies a persistent excitation condition, which may not be met with the standard industrial motion profile, as high-precision manufacturing requires, on the contrary, highly smooth motion.

Recent results incorporate the model uncertainty directly into the controller design. Polynomial regression is applied to approximate the uncertain inverse transfer function in [18] of an analog electronic circuit, which shows great robustness to parametric model uncertainty and measurement noise. The polynomial extrapolation method is extended to the prediction and compensation of the unknown disturbance in [19] for a timing-belt actuator, where a compensating control mechanism is presented to account for the prediction error of the model. The technique of Bayesian optimization is used for safe learning of controller parameters considering safety critical constraints in [20]–[22]. However, the common drawback of these methods is their lack of robustness to unmatched dynamic uncertainties, which limits the tracking performance, especially for high-order systems, due to the limited model order applicable for real-time application.

Robust control design approaches, such as \mathcal{H}_∞ design, have also received special attention due to the inherent robustness against both parametric and dynamic uncertainties. The robust inversion-based feedforward design method is presented in [23], [24] to directly account for and minimize the effect of dynamic uncertainty. This strategy is extended to a multiple-input and multiple-output problem in [25] with a mixed-sensitivity formulation. The controllers resulting from the classical unstructured \mathcal{H}_∞ synthesis have a full order of at least the model order plus the order of weights, and they rarely find their way into the industrial hardware. This can be addressed by the technique of non-smooth optimization presented in [26] and [27], which allows direct robust synthesis of fixed-structure controllers. However, it is clear that work on the robust synthesis of structured inverse feedforward control

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Haijia Xu, Christoph Hinze, and Alexander Verl are with the Institute for Control Engineering of Machine Tools and Manufacturing Units, University of Stuttgart, Stuttgart 70174, Germany (e-mail: haijia.xu@isw.uni-stuttgart.de; christoph.hinze@isw.uni-stuttgart.de; alexander.verl@isw.uni-stuttgart.de).

Andrea Iannelli is with the Institute for Systems Theory and Automatic Control, University of Stuttgart, Stuttgart 70569, Germany (e-mail: andrea.iannelli@ist.uni-stuttgart.de).

for feed drives is still scarce.

Our previous work introduced an velocity feedforward scheme based on regression trees (RTs) [28] to compensate for steady-state errors due to the load-varying elastic deformation. The feedforward relies on the numerical differentiation of the RTs, which can lead to large deviations in aggressive motion profiles due to their non-differentiable property.

This paper addresses the main shortcomings of our earlier work and others in the literature of feedforward control in two important directions concerning modeling and control design. First, we develop a hybrid model geared towards the use for robust feedforward design to improve the transient and steady-state tracking behavior simultaneously. The proposed hybrid model combines an analytical low-order approximation of the linear drive dynamics, and a data-driven Gaussian process (GP) model [29] of the output nonlinearity. Unlike the works that represent the entire system dynamics with GP models [30]–[32], our proposed approach approximates the dominant linear dynamics with an analytical model, which simplifies the learning task of the GP model to the static output nonlinearity with normalized problem scale. In addition, the flatness of the selected analytical model allows direct model inversion for feedforward control without the need for additional optimization [33], or inverse learning [34], [35]. Second, in contrast to the conventional exact model inversion [1], a modified inverse feedforward with fixed structure is proposed to account for model uncertainties. The parameterization of feedforward gains is formulated as a signal-based robust control problem with simultaneous consideration of multiple performance requirements, where the resulting design problem is solved using the structured μ synthesis technique presented in [26], [27]. In addition, guidelines on weight selection are provided to reduce the complexity of the control design for practitioners. The main contribution of the paper can be summarized as follows:

1. Hybrid modeling strategy of feed drives with particular focus on feedforward control, combining an analytical approximation of linear dynamics and a data-driven GP model of output nonlinearities.
2. Robust design procedure of modified feedforward gains using the structured μ synthesis technique to optimize multi-objective control performance under uncertainty in analytical and data-driven models.
3. Signal-based formulation of synthesis problem and practical guidelines for weight selection that limit the commissioning effort of feedforward gains to the selection of two hyperparameters.
4. Validated real-time capability, performance improvement and robustness to model errors on industrial hardware, with experimental data openly available in [36] for reproducibility and further analysis.

The rest of the paper is organized as follows: Sec. II introduces the industrial standard feedforward controls and the performance limiting assumptions. Sec. III proposes the hybrid modeling structure of feed drives, followed by an inversion-based feedforward design for tracking and disturbance compensation. Sec. IV proposes the robust synthesis framework

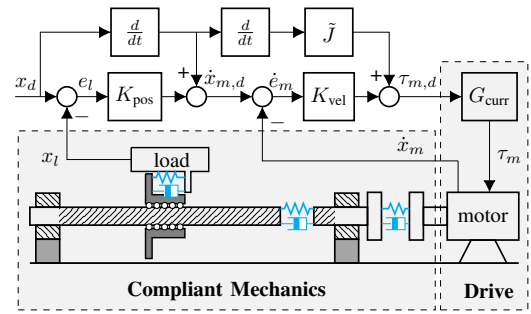


Fig. 1. Industrial cascaded control structure and mechanical properties of a feed drive with ball screw.

of feedforward gains, as well as guidelines for the weight selection to ensure industrial applicability. Sec. V presents an experimental validation of the proposed robust feedforward scheme on industrial hardware. Finally, Sec. VI gives the concluding remarks.

II. PROBLEM STATEMENT

Feed drives are an important motion generating part of machine tools converting the rotatory motion of a motor into a linear motion of the tool or table. The most common type of feed drives are ball screw drives due to their high stiffness, low friction and comparatively low cost, where the motor drives a screw spindle and the translational load side is connected by a chain of balls rolling between screw and the nut, as pictured in Fig. 1. Practically all industrial control platforms use a cascaded feedback control structure, consisting of a load-side proportional (P) position controller K_{pos} and a motor-side proportional-integral (PI) velocity controller K_{vel} , which determine the desired velocity $\dot{x}_{m,d}$ from position error e_l and desired torque $\tau_{m,d}$ from velocity error \dot{e}_m , respectively. Further, a PI current controller is used to control the current—and, hence, the motor torque—via pulse-width modulation. The closed-loop current control loop, named G_{curr} in Fig. 1, is typically by orders of magnitude faster than the mechanical behavior and the achievable frequency range of velocity and position controller [2]. Hence, for the remainder of this paper, we simplify $G_{curr}(s) \approx 1$. In addition, a differential feedforward control is used to compensate for the known behavior and allows for better error regulation through feedback. The velocity feedforward is used to cancel the tracking offset in constant velocity stages. The acceleration profile is converted to a torque feedforward term and compensates for the inertia \tilde{J} during the acceleration and deceleration. This control structure works well for stiff systems and is easy to parameterize as the control loops can be tuned sequentially, starting with the innermost current controller. However, for more dynamic motions or larger masses to be moved, the finite stiffness of the coupling, spindle, and nut leads to dynamic positioning errors as well as imperfectly manufactured parts, such as the spindle lead, which is subject to changes along the travel length of the feed drive.

The goal of this work is to achieve better output tracking of the commanded position x_d by improving the feedforward part in Fig. 1. The feedback control is assumed to be predefined

and is not changed. Although it might be beneficial to consider feedforward and feedback simultaneously, we decide not to do so here to ensure easier applicability in industrial practice, where the cascaded P-PI control structure is implemented in the frequency inverters and can hardly be changed in industrial applications, only parameterized. Also, the parameterization of feedback gains for multi-axis machines should account for the overall machine dynamics to synchronize the tracking behavior of all axes [37], which is not considered in this work. At the same time, the feedforward signal can be freely commanded externally from the CNC via the fieldbus system [38, §7].

Note that the standard velocity and acceleration feedforward controls in Fig. 1 perform the inversion of the inner motor control loop (from $\dot{x}_{m,d}$ to \dot{x}_m) and the mechanics (from $\tau_{m,d}$ to x_l), respectively. This relies on the fundamental rigid body assumptions, i.e.

1. The transfer function of velocity control loop has a constant magnitude of 1 for all frequencies.
2. The entire power train components are characterized by a rigid body with inertia \tilde{J} .

However, neglecting structural vibration modes and nonlinear characteristics of the mechanics results in limited output tracking performance [37]. Moreover, as the corresponding dynamics of the inner loop or mechanics change, e.g. due to changes in inertia, friction and other dynamics resulting from wear, aging or variations in lubrication over the machine's lifetime, the feedforward would compensate for the incorrect model [1]. This motivates the need for a more accurate feedforward strategy and a robust control design method to account for model uncertainties.

III. INVERSION-BASED FEEDFORWARD WITH HYBRID MODELING

This section proposes a combined analytical and data-driven modeling approach of the drive control system, followed by a feedforward control design based on the model inversion to improve the output tracking.

A. Hybrid Modeling Structure

In conventional feedforward design of feed drives, the motor torque is often chosen as the control input to account for the known dynamics of the plant or disturbance [39]. However, this requires a rather precise dynamics model of the entire compliant mechanics from motor torque to load position, which significantly increases the modeling effort.

The central idea of our modeling approach is to take the inner feedback loop as the first part of the model, and to use the commanded motor velocity $\dot{x}_{m,d}$ as the control signal. This modeling strategy shifts the objective of feedforward design from the inversion of the entire mechanical system, to the inversion of the inner control loop and the concatenated output mapping. The hybrid modeling, given in Fig. 2, assumes linear dynamics of the velocity control loop described by the analytical model G_0 , followed by a nonlinear output mapping captured by the data-driven model Φ .

The selected model structure offers two advantages that make it attractive from a practical point of view. On the one hand, in contrast to modeling the entire mechanics, taking the velocity control loop as the first part of the model reduces the sensitivity to plant variations and disturbances, allowing the corresponding dynamics to be described with a simple low-order analytical model and its corresponding uncertainty set with much less identification effort. On the other hand, as the dominant linear dynamics are captured by the analytical model, describing the remaining nonlinear output mapping is less demanding. This can be conveniently modelled as a static nonlinearity and identified with data-driven techniques such as Gaussian process (GP) regression.

B. Analytical Model of Velocity Control Loop

We use a linear reduced-order model to describe the dominant dynamics of velocity-controlled motor drive, namely to capture the first resonant mode. This model is based on the cascade control principle, which assumes that the velocity control loop of the motor drive operates on a much faster timescale than the mechanical dynamics. As such, the motor velocity loop is approximated as the transfer function from the desired velocity $\dot{x}_{m,d}$ to the actual velocity \dot{x}_m , given by

$$G_m(s) = \frac{\dot{X}_m(s)}{\dot{X}_{m,d}(s)} = \frac{\omega_0^2}{s^2 + 2D_0\omega_0s + \omega_0^2}. \quad (1)$$

where ω_0 represents the first resonant frequency and D_0 describes the damping ratio of the velocity loop. Also, the DC-gain $G_m(0)$ is chosen to be 1, as the velocity control loop has an integrating part in the controller. Thus, the analytical model G_0 (from $\dot{x}_{m,d}$ to x_m) is given by the velocity transfer function of the motor drive followed by an integrator, namely $G_0(s) = G_m(s)/s$.

Apart from the need for a good approximation of the dominant dynamics at low frequencies, the structure of the analytical model G_0 is chosen with a particular focus on the targeted feedforward design, i.e.

1. The model G_0 is selected to be flat.
2. The order of the model G_0 is limited to 3.

The flatness of the selected model simplifies the inversion-based feedforward design using smooth reference trajectories, even if the model inverse is not proper, see Sec. III-D. Moreover, limiting the model order to 3 has the practical motivation that CNC-guided motion is planned continuously up to the third derivative of the axis position (axis jerk). This motion profile will be used later to resolve the exact model inversion by explicitly using the known derivatives. Increasing the model order requires higher-order derivatives of the trajectory, which are not available in the standard CNC system [38, §5.6.2].

C. Data-driven Model of Compliant Mechanics

Following the linear dynamics model of the drive motor, the subsequent nonlinear output mapping Φ characterizes the nonlinear mechanics of the power train components, given by

$$x_l = \Phi(x_m) = \underbrace{x_m}_{=: \Phi_L} + \underbrace{(x_l - x_m)}_{=: \Phi_{NL}}. \quad (2)$$

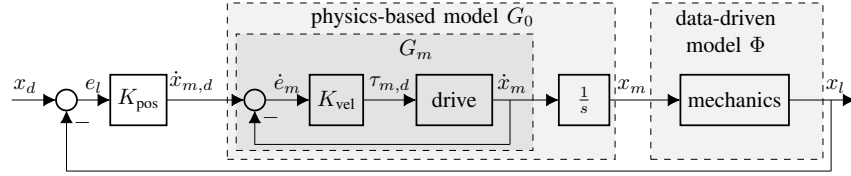


Fig. 2. Hybrid modeling of the feed drive control system for feedforward design.

This is further separated into a linear term Φ_L and a nonlinear term Φ_{NL} in addition, which have very different problem scales. The linear term Φ_L serves as the base model, and incorporates the prior knowledge that the drive train exhibits mostly a linear transmission behavior, affected by a secondary nonlinear distortion Φ_{NL} of much smaller magnitude. In contrast to learning the entire nonlinear mapping Φ containing different problem scales, this separation strategy simplifies the task of data-driven model to residual learning of Φ_{NL} by subtracting the linear base model Φ_L . Also, this additive representation simplifies the inversion-based feedforward in Sec. III-D, and allows the robust control design using the μ synthesis technique in Sec. IV-B.

The linear base model Φ_L represents the nominal transmission behavior of the powertrain components, namely the transmission ratio from rotational motion of the drive to axial motion of the load. The nonlinear distortion Φ_{NL} is observed to be patterned and periodic depending on the axis position and velocity (see Fig. 10 and [40]), due to the non-constant gear ratio resulting from the machining tolerances of the ball screw spindle, and the cyclical motion of the motor drive. This is typically approximated by parametric sinusoidal models with position and velocity dependent offsets, whose results rely heavily on expert knowledge of the parametric structure [41]. In contrast to this, the data-driven approach based on Gaussian process regression is applied in the following.

Consider the vector-valued input $\mathbf{x} = [x, \dot{x}]^\top$ consisting of the axis position and velocity, and the scalar-valued noisy output y_N , representing the measured nonlinear distortion Φ_{NL} subject to the Gaussian noise ε

$$y_{N,i} = \Phi_{NL}(\mathbf{x}_i) + \varepsilon_i \quad i = 1, \dots, n_D, \quad \varepsilon \sim \mathcal{N}(0, \sigma_N^2). \quad (3)$$

Then the posterior distribution under the Gaussian prior and likelihood is also Gaussian [29]. Conditioning on the training data set $\mathbf{X} = [\mathbf{x}_1, \dots, \mathbf{x}_{n_D}]$ and $\mathbf{y} = [y_{N,1}, \dots, y_{N,n_D}]$ of length n_D , the prediction of $\Phi_{NL}(\mathbf{x})$ at an arbitrary test input \mathbf{x} is given by the posterior mean and variance

$$\text{mean} = m(\mathbf{x}) + k(\mathbf{x}, \mathbf{X})^\top \underbrace{(k(\mathbf{X}, \mathbf{X}) + \sigma_N^2)^{-1}(\mathbf{y} - m(\mathbf{X}))}_{=: \beta} \quad (4)$$

$$\text{var} = k(\mathbf{x}, \mathbf{x}) - k(\mathbf{x}, \mathbf{X})^\top (k(\mathbf{X}, \mathbf{X}) + \sigma_N^2)^{-1} k(\mathbf{x}, \mathbf{X}). \quad (5)$$

The mean function $m(\cdot)$ incorporates the prior knowledge of the trend in the data and can be used to improve the extrapolation behavior [29]. This is set to 0 as we are only concerned with the interpolation behavior within the predefined operational space. The kernel function $k(\cdot, \cdot)$ provides a

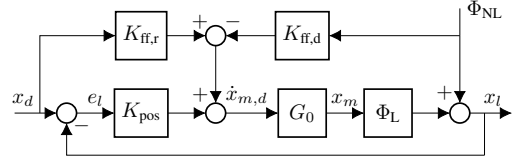


Fig. 3. Control structure with hybrid feedforward compensation.

similarity measure over function values in the input space, and the squared exponential kernel is used for continuous approximation, given by

$$k_{SE}(\mathbf{x}, \mathbf{x}') = \sigma_S^2 \exp\left(-\sum_{j=1}^{n_x} \frac{(x_j - x'_j)^2}{2l_j^2}\right) \quad (6)$$

where n_x is the number of inputs, σ_S^2 is the signal variance that determines the average distance of the nonlinear function $\Phi_{NL}(\cdot)$ from its mean, and l_j is the length scale that captures the correlation of neighboring points along a given axis in the input space.

D. Feedforward Control with Model Inversion

Based on the separation strategy in Eq. (2), the nonlinear output mapping Φ can be further described as a linear transfer function Φ_L influenced by an additional disturbance Φ_{NL} . The corresponding control structure with hybrid feedforward for the tracking of reference x_d and the rejection of disturbance Φ_{NL} is shown in Fig. 3.

The linear transfer function Φ_L , which determines the nominal transmission ratio of the powertrain, has a magnitude of 1, as discussed in Sec. III-C. It is thus neglected in the following for simplicity. The additive disturbance term Φ_{NL} is approximated by GP model, which takes the desired reference as input for prediction rather than measurements to avoid feedback loops. In the frequency domain, the achieved output load position x_l with the desired reference x_d is given by

$$x_l = (1 + G_0 K_{pos})^{-1} G_0 (K_{ff,r} + K_{pos}) x_d + (1 + G_0 K_{pos})^{-1} (1 - G_0 K_{ff,d}) \Phi_{NL}, \quad (7)$$

with the control law

$$u = \underbrace{K_{ff,r} x_d - K_{ff,d} \Phi_{NL}}_{\text{feedforward}} + \underbrace{K_{pos} (x_d - x_l)}_{\text{feedback}}, \quad (8)$$

where K_{pos} is the proportional position controller inherent in the drive control system, $K_{ff,r}$ and $K_{ff,d}$ are the feedforward controllers that are to be designed for trajectory tracking and disturbance rejection, respectively.

359 A standard approach adopted in practice to design $K_{\text{ff},r}$
 360 and $K_{\text{ff},d}$ is to use the so-called exact model inverse. That is,
 361 assuming the transfer function G_0 is exact, the feedforward
 362 controllers can be chosen as the inverse of the model for
 363 tracking and disturbance rejection

$$K_{\text{ff},r} = K_{\text{ff},d} = G_0^{-1} = \frac{s^3 + 2D_0\omega_0s^2 + \omega_0^2s}{\omega_0^2}. \quad (9)$$

364 If we assume in addition that the map Φ_{NL} is also known,
 365 this inverse feedforward achieves exact output tracking, i.e.
 366 by substituting the feedforward law of Eq. (9) into Eq. (7),
 367 we obtain $x_l = x_d$.

368 The respective feedforward control laws for tracking and
 369 disturbance rejection can be expressed in the time domain as

$$u_{\text{ff},r} = \frac{1}{\omega_0^2} \ddot{x}_d + \frac{2D_0}{\omega_0} \dot{x}_d + \dot{x}_d, \quad (10)$$

$$u_{\text{ff},d} = \frac{1}{\omega_0^2} \ddot{\Phi}_{\text{NL}} + \frac{2D_0}{\omega_0} \dot{\Phi}_{\text{NL}} + \Phi_{\text{NL}}, \quad (11)$$

370 with the desired velocity \dot{x}_d , acceleration \ddot{x}_d and jerk \dddot{x}_d of
 371 the reference signal. Similarly, $\dot{\Phi}_{\text{NL}}$, $\ddot{\Phi}_{\text{NL}}$ and $\ddot{\Phi}_{\text{NL}}$ represent
 372 the first, second and third time derivatives of the nonlinear
 373 distortion, respectively. The GP prediction only takes the
 374 desired trajectory as input for feedforward control, i.e. $x = x_d$
 375 and $\dot{x} = \dot{x}_d$, to avoid introducing additional feedback loops.

376 Furthermore, for the computation of time derivatives of the
 377 GP model in Eq. (11), we neglect higher-order derivatives of
 378 the desired trajectory and consider $\dot{x}_d \approx 0$, which basically
 379 limits the prediction of the derivatives to the constant veloc-
 380 ity phase. Exact calculation without neglecting higher-order
 381 derivatives can, potentially, improve the transient behavior
 382 even further. However, the practical motivation is that without
 383 this simplification, the fourth time derivative of the reference
 384 trajectory \dddot{x}_d would be required to compute the third time
 385 derivative $\ddot{\Phi}_{\text{NL}}(x_d, \dot{x}_d)$, which is not available in the standard
 386 industrial numerical control system [38, §5.6.2].

387 Therefore, considering the two inputs x and \dot{x} of the GP
 388 model with $\ddot{x} \approx 0$, the time derivatives are given by

$$\begin{aligned} \dot{\Phi}_{\text{NL}}(x, \dot{x}) &= \frac{\partial \Phi_{\text{NL}}}{\partial x} \dot{x} + \underbrace{\frac{\partial \Phi_{\text{NL}}}{\partial \dot{x}} \ddot{x}}_{=0} \\ \ddot{\Phi}_{\text{NL}}(x, \dot{x}) &= \frac{\partial^2 \Phi_{\text{NL}}}{\partial x^2} \dot{x}^2 + \underbrace{\frac{\partial^2 \Phi_{\text{NL}}}{\partial x \partial \dot{x}} \dot{x} \ddot{x}}_{=0} + \underbrace{\frac{\partial \Phi_{\text{NL}}}{\partial x} \ddot{x}}_{=0} \\ \ddot{\Phi}_{\text{NL}}(x, \dot{x}) &= \frac{\partial^3 \Phi_{\text{NL}}}{\partial x^3} \dot{x}^3 + \underbrace{\frac{\partial^3 \Phi_{\text{NL}}}{\partial x^2 \partial \dot{x}} \dot{x} \dot{x}^2}_{=0} + 2 \underbrace{\frac{\partial^2 \Phi_{\text{NL}}}{\partial x^2} \dot{x} \ddot{x}}_{=0}, \end{aligned} \quad (12)$$

389 where the derivatives of the GP model with respect to its inputs
 390 can be obtained by the chain rule according to Eq. (4).

391 In practice, the nominal model G_0 may not exactly repre-
 392 sent the true plant, especially at high frequencies. Moreover,
 393 the GP model cannot fully capture the characteristics of
 394 the disturbance term Φ_{NL} and the prediction is subject to
 395 uncertainties captured by the variance in Eq. (5). Also, the
 396 relevant frequency ranges of tracking and disturbance rejection
 397 are different. In contrast to the same exact inverse for $K_{\text{ff},r}$
 398 and $K_{\text{ff},d}$ in Eq. (9), it is thus advantageous to select the

399 feedforward gains separately [42]. This leads to the need for
 400 a robust multi-objective feedforward design method that seeks
 401 to achieve the best possible performance over the possible
 402 uncertainties for tracking and disturbance rejection.

IV. ROBUST FEEDFORWARD SYNTHESIS UNDER MIXED UNCERTAINTIES

403 This section proposes a robust feedforward design method
 404 via structured μ -synthesis to optimize the robust performance
 405 of the inversion-based feedforward controller described in
 406 Sec. III. In addition, weight selection guidelines are presented
 407 to give practitioners an intuitive insight into the trade-offs of
 408 the robust design.
 409
 410

A. Modelling of Uncertainties

411 For the inverse feedforward control design in Sec. III, the
 412 feed drive control system is represented by a hybrid model:
 413 the analytical model G_0 of drive dynamics approximated by
 414 a second order lag term in Eq. (1) with an integrator, and the
 415 data-driven model Φ_{NL} of mechanical transmission represented
 416 by GP regression in Eq. (4). Both of them are still subject to
 417 uncertainties, namely the complex dynamic uncertainty of G_0
 418 and the real parametric uncertainty of the GP model.
 419

420 Consider the set Π of all possible plants under uncertainty,
 421 the complex dynamic uncertainty of the nominal analytical ap-
 422 proximation can be captured by the multiplicative uncertainty
 423 model in the frequency domain as

$$G_p(j\omega) = G_0(j\omega)(1 + W(j\omega)\Delta_c(j\omega)), \quad (13)$$

424 where $G_p \in \Pi$ describes the possible uncertain plant, G_0 is
 425 the nominal model and $\Delta_c \in \mathbb{C}$ is the normalized complex
 426 uncertainty with $|\Delta_c| < 1$. The weight W represents the
 427 variation of the relative model uncertainty in the frequency
 428 domain, and its magnitude satisfies

$$|W(j\omega)| \geq l_m(\omega) = \max_{G_p \in \Pi} \left| \frac{G_p(j\omega) - G_0(j\omega)}{G_0(j\omega)} \right|, \quad \forall \omega. \quad (14)$$

429 Here, l_m captures the largest possible magnitude of the relative
 430 model uncertainty over frequencies. The uncertainty weight
 431 W determines the size of the considered uncertainty set, and
 432 must be chosen to have a greater magnitude than l_m , to ensure
 433 that all possible relative uncertainties are included within
 434 the uncertainty model of Eq. (13). The weight W is often
 435 chosen as a high-pass filter [42], corresponding to the fact that
 436 the nominal low-order approximation G_0 mainly captures the
 437 dynamics at low frequencies and has a larger error at higher
 438 frequencies.

439 In addition, the uncertainty of the disturbance prediction
 440 Φ_{NL} by the GP model is described by an additive parametric
 441 uncertainty model with prediction error bounds. We use d_0 as
 442 the nominal disturbance term predicted by the GP model and
 443 d as the true disturbance Φ_{NL} . This is given by

$$|d - d_0| \leq 3\sigma, \quad \forall \omega. \quad (15)$$

444 The uncertain disturbance d is described by the 3σ confidence
 445 region around the mean GP prediction d_0 . This can be repre-
 446 sented as an additive uncertain disturbance model, given by

$$d = d_0 + 3\sigma \cdot \Delta_r, \quad (16)$$

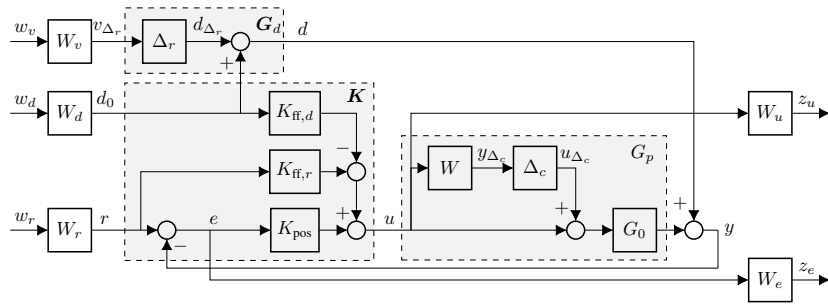


Fig. 4. Signal-based robust performance problem for controller synthesis.

447 with normalized parametric uncertainty $\Delta_r \in \mathbb{R}$ and $|\Delta_r| < 1$.
 448 Besides, the variance σ is estimated in a conservative way
 449 by the maximum variance of the GP model over the entire
 450 input space. Noticeably, the frequency-varying uncertainty
 451 quantification is not considered here due to numerical diffi-
 452 culties. The practical problem is that the secondary nonlinear
 453 distortion Φ_{NL} has a rather small magnitude compared to its
 454 input vector. In our case, the identification of the investigated
 455 transfer function, if possible, has a relevant magnitude of about
 456 -65 dB, which makes the frequency domain GP model very
 457 sensitive to measurement noise and numerical errors.

458 B. Signal-based Robust Feedforward Synthesis

459 The central idea of the robust feedforward control synthesis
 460 is to seek for the best achievable performance over the set
 461 of possible uncertainties [43]. In contrast to the exact inverse
 462 feedforward given in Eq. (9), the modified inverse feedforward
 463 is used to account for model uncertainties, especially at high
 464 frequencies. The modified feedforward structure is given by

$$465 K_{\text{ff},i} = F_{c,i} G_{0,i}^{-1} = \frac{\frac{1}{\omega_{0,i}^2} s^3 + \frac{2D_{0,i}}{\omega_{0,i}} s^2 + s}{(T_{c,i} s + 1)^3}, \quad (17)$$

466 where the subscript i denotes r and d for reference tracking
 467 and disturbance compensation, respectively. The lag term
 468 $F_{c,i} = 1/(T_{c,i} s + 1)^3$ is introduced to capture the band limit
 469 of the feedforward gain and to restrict the model inversion
 470 to frequency regions of low uncertainty. Equivalently, the
 471 crossover frequency can be calculated as $f_{c,i} = 1/(2\pi \cdot T_{c,i})$
 472 in Hz.

473 Also, unlike the exact inverse in Eq. (9), whose parameters
 474 are determined by the identification in the frequency domain,
 475 the parameters $\omega_{0,i}$, $D_{0,i}$ and $T_{c,i}$ of feedforward controllers
 476 are determined by the robust synthesis framework for robust
 477 performance optimization. In addition, although the feedfor-
 478 ward gains for tracking and disturbance rejection take the same
 479 structure of Eq. (17), the corresponding control parameters are
 480 synthesized independently as their relevant frequency ranges
 481 are different.

482 The synthesis of robust feedforward controllers for trajec-
 483 tory tracking and disturbance rejection is formulated as a
 484 signal-based problem [42, §9.3.6], which is very general and
 485 appropriate for multivariable problems considering multiple
 486 performance objectives simultaneously, as shown in Fig. 4.

487 The transfer functions G_p and G_d represent the uncertain
 plant and disturbance model, $K_{\text{ff},r}$ and $K_{\text{ff},d}$ are the two

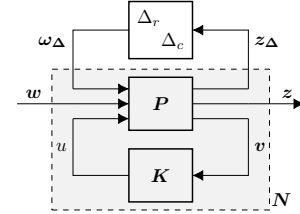


Fig. 5. Generalized robust synthesis interconnection.

feedforward controllers that are to be synthesized, K_{pos} is
 the proportional position controller with fixed gain inherent
 in the original control system. The input weights W_v , W_d
 and W_r represent the mapping from the exogenous signals
 to the corresponding physical signals, namely the parametric
 uncertainty of the GP, the predicted nominal disturbance, and
 the reference trajectory. The output weights W_u and W_e
 specify the desired performance requirements in terms of the
 control effort and the control error, respectively.

For the controller synthesis, the signal-based interconnec-
 tion in Fig. 4 can be transformed into the generalized robust
 synthesis structure of Fig. 5 by introducing

$$488 \mathbf{w} = \begin{bmatrix} w_v \\ w_d \\ w_r \end{bmatrix}, \quad \mathbf{z} = \begin{bmatrix} z_u \\ z_e \end{bmatrix}, \quad \mathbf{v} = \begin{bmatrix} d_0 \\ r \\ y \end{bmatrix}, \quad u = u, \quad (18)$$

489 where $\Delta = \text{diag}[\Delta_r, \Delta_c]$ is the uncertainty set with real
 490 and complex blocks, \mathbf{P} is the generalized plant, and \mathbf{K}
 491 is the generalized controller; \mathbf{v} are the measured outputs of
 492 the general plant and u is the control input consisting of
 493 the feedforward and feedback parts; $\omega_{\Delta} = [d_{\Delta_r}, u_{\Delta_c}]^{\top}$
 494 and $\mathbf{z}_{\Delta} = [v_{\Delta_r}, y_{\Delta_c}]^{\top}$ are the uncertain inputs and outputs,
 495 respectively.

496 The generalized plant \mathbf{P} is given by a transfer function
 497 matrix as

$$498 \begin{bmatrix} v_{\Delta_r} \\ y_{\Delta_c} \\ z_u \\ z_e \\ d_0 \\ r \\ y \end{bmatrix} = \underbrace{\begin{bmatrix} 0 & 0 & W_v & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & W \\ 0 & 0 & 0 & 0 & 0 & W_u \\ -W_e & -W_e G_0 & 0 & -W_e W_d & W_e W_r & -W_e G_0 \\ 0 & 0 & 0 & W_d & 0 & 0 \\ 0 & 0 & 0 & 0 & W_r & 0 \\ 1 & G_0 & 0 & W_d & 0 & G_0 \end{bmatrix}}_{=: \mathbf{P}} \begin{bmatrix} d_{\Delta_r} \\ u_{\Delta_c} \\ w_v \\ w_d \\ w_r \\ u \end{bmatrix}. \quad (19)$$

509 The generalized controller \mathbf{K} with fixed structure reads

$$u = \underbrace{\begin{bmatrix} -K_{ff,d} & K_{ff,r} + K_{pos} & -K_{pos} \end{bmatrix}}_{=: \mathbf{K}} \begin{bmatrix} d_0 \\ r \\ y \end{bmatrix}, \quad (20)$$

510 where K_{pos} is the proportional feedback controller, $K_{ff,r}$ and
511 $K_{ff,d}$ are the feedforward controllers of Eq. (17) for tracking
512 and disturbance rejection, respectively.

513 To analyze the robust performance of the uncertain system,
514 the interconnection of Fig. 5 can be transformed into the $\mathbf{N}\Delta$
515 structure by relating the transfer function matrix \mathbf{N} (from
516 $[\omega_{\Delta}^T, \omega^T]^T$ to $[z_{\Delta}^T, z^T]^T$) to \mathbf{P} and \mathbf{K} by a lower linear
517 fractional transformation

$$\mathbf{N} = \mathcal{F}_l(\mathbf{P}, \mathbf{K}) = \mathbf{P}_{11} + \mathbf{P}_{12}\mathbf{K}(\mathbf{I} - \mathbf{P}_{22}\mathbf{K})^{-1}\mathbf{P}_{21}, \quad (21)$$

518 which can be further rearranged into the $\mathbf{M}\Delta$ structure for
519 robust stability analysis, with the upper left block of \mathbf{N}
520 representing the transfer function matrix \mathbf{M} from the uncertain
521 inputs ω_{Δ} to the uncertain outputs z_{Δ} .

522 Nominal and robust stability are the prerequisites for robust
523 performance. Designing the feedback controller such that the
524 system remains stable under uncertainties, as discussed in
525 Sec. II, is not the focus of this paper. In the following we
526 assume that the stability conditions are satisfied and focus
527 on the robust performance optimization by synthesis of the
528 feedforward gains.

529 Both robust stability and performance problems can be ad-
530 dressed using the technique of μ -analysis [42]. The structured
531 singular value (SSV) of the transfer function matrix \mathbf{M} , in
532 terms of the normalized uncertainty set Δ with maximum
533 singular value $\bar{\sigma}(\Delta)$ less than one, is given by [42, §8.8]

$$\mu_{\Delta}(\mathbf{M}) = \frac{1}{\min\{k_m \mid \det(\mathbf{I} - k_m\mathbf{M}\Delta) = 0, \bar{\sigma}(\Delta) \leq 1\}}. \quad (22)$$

534 The inverse of the SSV value $\mu_{\Delta}(\mathbf{M})$ determines the smallest
535 positive value that gives a singular matrix $\mathbf{I} - k_m\mathbf{M}\Delta$, which
536 corresponds to an unstable interconnection between $k_m\mathbf{M}$
537 and Δ . In other words, the inverse of $\mu_{\Delta}(\mathbf{M})$ indicates the
538 maximum tolerable increase of the uncertainty set Δ , before
539 the closed-loop control system becomes unstable. Thus, the
540 robust stability condition reads

$$\mu_{\Delta}(\mathbf{M}) < 1, \quad (23)$$

541 yielding a robust stabilization of the plant \mathbf{P} by the controller
542 \mathbf{K} subject to any uncertainties within the uncertainty set Δ .

543 Furthermore, by introducing the extended block structure
544 $\Delta_{\text{ext}} = \text{diag}[\Delta, \tilde{\Delta}]$ with the actual uncertainty set Δ and
545 a normalized full complex uncertainty $\tilde{\Delta}$ [42, §8.10.1], the
546 robust performance condition of the interconnection of Fig. 5
547 can be transformed into the robust stability condition of the
548 extended $\mathbf{N}\Delta_{\text{ext}}$ structure, given by

$$\mu_{\Delta_{\text{ext}}}(\mathbf{N}) < 1, \quad (24)$$

549 which corresponds to the satisfaction of the control perfor-
550 mance specifications subject to the uncertainty set Δ , even in
551 the worst case. The robust performance synthesis then amounts

to designing a controller \mathbf{K} of Eq. (20) that minimizes the SSV
552 value $\mu_{\Delta_{\text{ext}}}(\mathbf{N})$, i.e.

$$\min_{\mathbf{K}} \mu_{\Delta_{\text{ext}}}(\mathbf{N}). \quad (25)$$

554 Although the search for the fixed-structure controller \mathbf{K}
555 of Eq. (20) that satisfies the condition of Eq. (24) has not
556 been fully solved, locally optimal solutions can be found by
557 combining the μ -analysis and the structured \mathcal{H}_{∞} -synthesis.
558 The main idea is to iterate between the estimation of the
559 upper bound of μ via D-scaling (D-step) and the synthesis
560 of a structured \mathcal{H}_{∞} controller for the scaled problem (K-step)
561 using the non-smooth optimization technique [26], [27]. In
562 addition, to account for the real parametric uncertainty, the G-
563 scaling can be used to obtain a less conservative estimate of
564 the upper bound [44]. The DGK-iteration with fixed-structure
565 \mathcal{H}_{∞} -synthesis to solve problem (25) is available as `musyn`
566 program in MATLAB's Robust Control Toolbox.

C. On the Weight Selection

567 The weights of the signal-based robust control problem
568 represent the known or expected frequency content of the
569 signals, and specify the desired performance requirements in
570 terms of control input and control error. We presented in our
571 previous work [45] a two-step design approach of the weight
572 selection for the signal-based robust control problem, i.e.

- 573 1. Map the exogenous signals to the physical signals based
574 on the measurement.
- 575 2. Define the performance requirements by selecting two
576 hyperparameters.

577 The practical motivation for this design procedure is to limit
578 the tuning complexity and to allow even non-specialists to
579 use the proposed robust synthesis framework for feedforward
580 design with limited commissioning effort, summarized below.
581

Step 1: Information extraction from the measurement

582 The weights of the exogenous inputs are set according
583 to the expected magnitudes of the physical signals: The
584 reference weight W_r is set to the expected maximum reference
585 change within the working space; the disturbance weight W_d
586 takes maximum magnitude of the nominal disturbance d_0 ;
587 the parametric uncertainty weight W_v represents the error bound
588 of the GP model and is set to $W_v = 3\sigma$.
589

590 The dynamic uncertainty weight W represents the uncer-
591 tainty variation of the analytical model G_0 over frequencies.
592 As defined in Eq. (14), this is defined as a high-pass filter,
593 since the low-order approximation is less accurate at high
594 frequencies. We thus define its inverse as

$$W^{-1} = \frac{(s/M^{1/n_W} + \omega_B)^{n_W}}{(s + \omega_B A^{1/n_W})^{n_W}}. \quad (26)$$

595 The weight parameters are determined graphically from the
596 measured frequency response functions according to the con-
597 dition of Eq. (14), where n_W is the filter order determining
598 the slope, ω_B is the crossover frequency where the relative
599 uncertainty exceeds 1, and $M > 1$, $A < 1$ are the asymptote
600 at high and low frequencies, respectively.

601 The weight W_u describes the expected frequency content of
 602 the control signal and avoids input saturation. This is defined
 603 as a first order low-pass filter, given by

$$W_u = \frac{s/M_u + \omega_{B,u}}{s + \omega_{B,u}A_u}, \quad (27)$$

604 where a larger magnitude of W_u implies a smaller expected
 605 control action. Also, the parameters can be determined graph-
 606 ically in a similar way to Eq. (26), based on the measured
 607 frequency response function from the desired reference r to
 608 the control signal u in the standard control loop.

609 Step 2: Definition of performance requirements

610 In addition to the weights mentioned above, which are
 611 determined from the measurement, the performance weight
 612 W_e defines the required control performance with respect to
 613 the control error e , which is chosen by the designer. This is
 614 defined as a low-pass filter, i.e.

$$W_e = \frac{s/M_e + \omega_{B,e}}{s + \omega_{B,e}A_e}, \quad (28)$$

615 where a larger magnitude of W_e implies a smaller error
 616 tolerance. Due to the integrator of G_0 , we have $A_e = 0$
 617 inherently. However, the low frequency asymptote A_e is still
 618 set to a small value to avoid numerical errors [42, §2.7.3]. The
 619 remaining two hyperparameters $\omega_{B,e}$ and M_e are determined
 620 by the designer to trade-off between the expected bandwidth
 621 and the attenuation of high frequency oscillations. A larger
 622 value of the desired bandwidth $\omega_{B,e}$ results in lower tracking
 623 error at low frequency, but inevitably increases the peak M_e
 624 and the sensitivity to high frequency oscillations.

625 V. VALIDATION

626 The proposed robust feedforward control scheme has been
 627 validated experimentally on an industrial feed drive. For re-
 628 producibility and further analysis, the experimental data are
 629 openly available in [36].

630 A. Experimental Setup and Computational Requirements

631 The experimental setup consists of the x-axis of a five-axis
 632 milling machine, shown in Fig. 6. The motor is a Rexroth
 633 MS2N03-D0BYN with a rated torque of 0.68 Nm, maximum
 634 torque of 6.8 Nm and a rated velocity of 5700 1/min. The load
 635 (namely the z- and b-axis) weighs 150 kg and is driven on the
 636 Franke TSL06U ball screw linear table, which has a spindle
 637 lead of 5 mm and an effectively reachable length of 0.36 m.
 638 The motor is controlled with Rexroth ctrlX DRIVE coupled
 639 with Beckhoff TwinCAT 3 system for real-time control. All
 640 parts of the feedforward control, including the GP prediction,
 641 are implemented in PLC code on the PC-based TwinCAT 3
 642 real-time control system with a sampling rate of 1 kHz.

643 The most computationally intensive part of the feedforward
 644 scheme is the evaluation of the GP model for the distur-
 645 bance compensation of Eq. (11), which requires three times
 646 evaluation of the GP derivatives of Eq. (12). To obtain fast
 647 approximate prediction for the real-time control, the nearest
 648 neighbor approach [46] is used to approximate the full GP

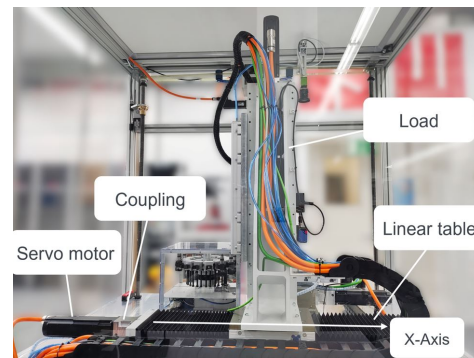


Fig. 6. The test bench used for validation.

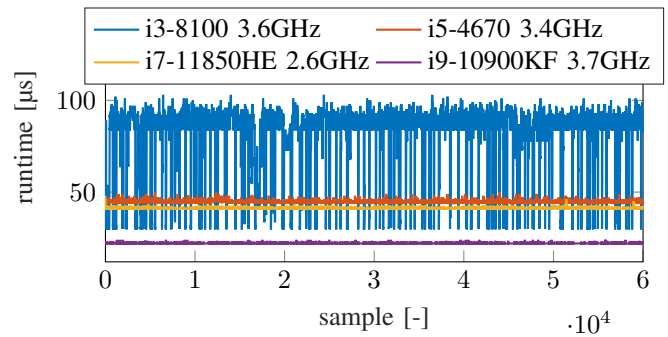


Fig. 7. Runtime of feedforward scheme with GP model on industrial PCs.

649 prediction of Eq. (4). The basic idea is that the GP kernels only
 650 determine the prediction locally, and the data points closest to
 651 the test input are the most informative. At each prediction
 652 step, the closest points \mathbf{X}^* with predefined box constraints
 653 are searched along each axis of the input, which can be easily
 654 implemented by index searching. The local approximation
 655 of the full GP prediction in Eq. (4) is then computed by
 656 multiplying $k(\mathbf{x}, \mathbf{X}^*)^\top$ by the coefficients β^* corresponding
 657 to \mathbf{X}^* . The sizes of the box constraints are determined by
 658 requiring a remaining accuracy of 99% compared to the full
 659 prediction, resulting in constraints of ± 20 mm in position and
 660 ± 10 mm/s in velocity. Such dimensions allow a good balance
 661 between computational effort and prediction accuracy.

662 The runtime of feedforward with GP prediction (single
 663 core performance) is measured in the TwinCAT 3 system on
 664 different CPUs, given in Fig. 7 and Table I. For example, on an
 665 i5-4670 CPU, the mean and maximum execution times for the
 666 compensation scheme are 45 and 53 μ s. Even with the weakest
 667 i3-8100 CPU in the test, the maximum runtime is 108 μ s,
 668 which is only 10% of the sampling time. Using vectorized code
 669 (SIMD instructions on the processor) could speed this up even
 670 more. In addition, a total memory of 29.1 kB is required to
 671 store the prediction parameter β of Eq. (4) in double precision.
 672 This illustrates the real-time capability and the small memory
 673 footprint of the compensation scheme.

674 B. Identification of Hybrid Model

675 The analytical model G_0 of the velocity control loop is
 676 identified using least-squares by comparing the measured and
 677 modelled frequency response functions (FRFs) [47, §9.9.1].

TABLE I
COMPUTATION TIME OF GP-BASED FEEDFORWARD SCHEME ON
DIFFERENT CPUs WITH SAMPLING TIME 1 MS.

CPU time	i3-8100	i5-4670	i7-11850HE	i9-10900KF
mean [μ s]	81	45	41	22
maximum [μ s]	108	53	48	24

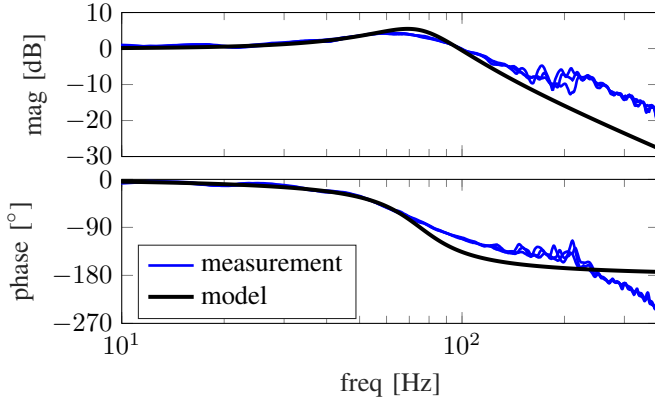


Fig. 8. Identified analytical model G_m of velocity loop without integrator.

To measure the FRFs, sinusoidal velocity sweeps are used with linearly increasing frequency $f \in [1, 400]$ Hz. An offset velocity of 10 mm/s is added to reduce the influence of stiction friction. Also, the FRFs are measured at different start positions $x_{l,0} \in \{0, 150, 300\}$ mm to capture the position-varying dynamics. The local rational model (LRM) method [48] is used to estimate the FRFs with a model order of 2 and a window length of 101.

The identified PT₂ model G_m of Eq. (1) (from $\dot{x}_{m,d}$ to \dot{x}_m , with $\omega_0 = 472.8$ rad/s and $D_0 = 0.28$) of the velocity loop without integrator is shown in Fig. 8. The corresponding multiplicative uncertainty, the largest possible magnitude of the relative model uncertainty l_m , and the selected uncertainty weight W are shown in Fig. 9. The uncertainty weight W is selected using the strategy introduced in Sec. IV-C with $n_W = 4$, $\omega_B = 2\pi \cdot 130$, $M = 15$ and $A = 0.11$, which has a larger magnitude than l_m to include all possible relative uncertainties over frequencies, see also Eq. (14). The relative uncertainty exceeds 1 at about 140 Hz, indicating that the low-order analytical model only captures the dynamics in the lower frequency range and deviates more than 100% at frequencies greater than 140 Hz. To capture the high frequency dynamics more accurately than the analytical model of Eq. 1, it is necessary to increase the model order. This would require motion profiles smoother than the jerk-limited trajectory, which, however, are not available in the standard numerical control system [38, §5.6.2].

The nonlinear distortion $\Phi_{NL} = x_l - x_m$ (c.f. Eq. (2)) is measured over the workspace at the commanded velocity $v_d \in [110, 210]$ mm/s with a grid of 10 mm/s. Fig. 10 shows the periodic pattern of the measured Φ_{NL} depending on the axis position and velocity, which is then captured by the GP regression model. The variance of the measurement noise is set as the square of the maximum relative error of the linear

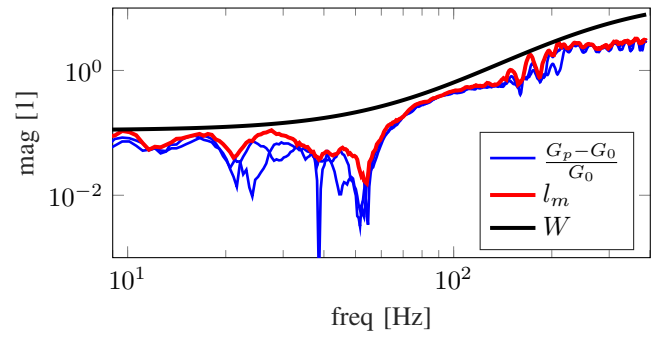


Fig. 9. Relative uncertainty and uncertainty weight of analytical model.

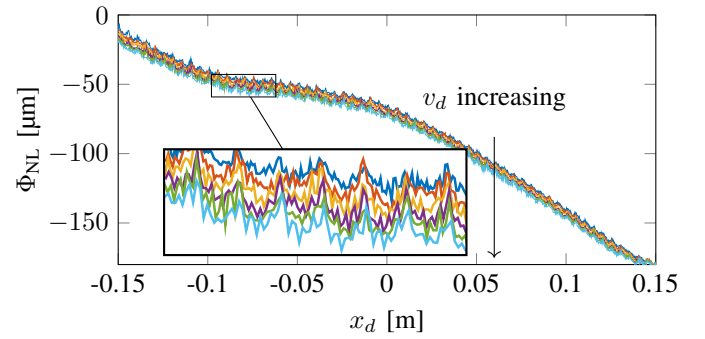


Fig. 10. Measured nonlinear distortion Φ_{NL} for different velocities.

encoder with $\sigma_N^2 = (5 \cdot 10^{-7})^2$. The signal variance is estimated according to the variance of the measured Φ_{NL} , which takes $\sigma_S^2 = (3 \cdot 10^{-5})^2$. A reasonable smoothness of the input space and a good prediction result are achieved with the length scale parameters $l_1 = 0.0015$ and $l_2 = 0.005$, which are chosen iteratively, and can also be estimated by likelihood maximization or cross validation [29, §5.4].

The validation on the test bench is performed with finer grids of 5 mm/s at unseen operating velocities to test the generalization capability of the model. The normalized validation result of the GP regression model at $\dot{x}_d = 175$ mm/s in the interval of 150 mm is shown in Fig. 11. Overall, a high coefficient of determination $R^2 = 97\%$ between measurement and prediction and a mean-absolute error of 1.13 μ m is obtained. A major advantage of the GP model over parametric approaches is the high degree of adaptability to the unseen

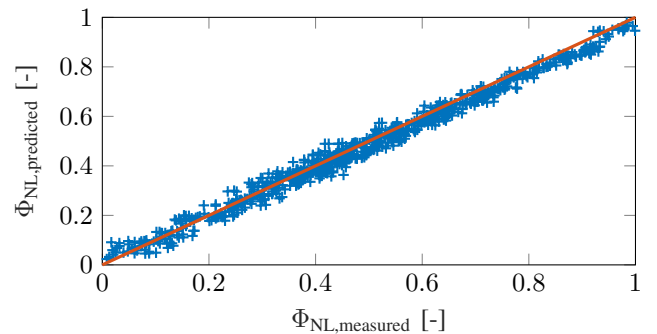


Fig. 11. Validation of normalized GP prediction on the test bench.

operating condition. In addition, if significant prediction error occurs, the measured data during the machine operation can be stored to update the GP parameters, and thus adapt the compensation scheme to new operating conditions. This might be the case due to wear during the lifetime of the feed drive.

C. Feedforward Control Design

The feedback controller of Fig. 1 remains the same and the following feedforward controllers are compared in the validation:

- baseline: The standard velocity and acceleration feedforward control given in Sec. II.
- exact inverse: The exact inverse feedforward control of the hybrid model given in Sec. III-D.
- robust inverse: The robust feedforward control with modified inverse of the hybrid model given in Sec. IV-B.

The gains of the exact inverse feedforward of Eq. (9) take directly the model parameters identified in the frequency domain, given in Sec. V-B and Table II. The robust parameterization of the modified inverse feedforward of Eq. (17) is performed based on the two-step approach introduced in Sec. IV-C. The selected weights determined from the measurement are: $W_d = 2 \cdot 10^{-4}$, $W_r = 0.36$, $W_v = 2.6 \cdot 10^{-6}$ and $W_u = (0.015s + 0.1257)/(s + 0.01)$. The performance weight is set to $W_e = (0.8s + 62.8)/(s + 0.00628)$ by requiring a sensitivity peak of $M_e = 1.25$ and a crossover frequency of $\omega_{B,e} = 10 \cdot 2\pi$, which gives a good balance between low frequency tracking and high frequency damping. The low frequency asymptote is set to $A_e = 10^{-4}$ to avoid numerical problems. The resulting peak μ value is $0.689 < 1$, indicating the satisfaction of robust performance requirements, and the corresponding feedforward gains are collected in Table II.

TABLE II
CONTROLLER PARAMETERS OF EXACT AND MODIFIED ROBUST INVERSE FOR HYBRID MODEL.

feedforward gains	$\omega_{0,i}$ [rad/s]	$D_{0,i}$ [-]	$f_{c,i}$ [Hz]
exact inverse $K_{ff,r}, K_{ff,d}$	472.8	0.28	-
robust inverse $K_{ff,r}$ $K_{ff,d}$	331.1 472.3	0.38 0.37	18.6 50.4

The modified robust inverse introduces band limitation for the feedforward control, which is implemented separately for reference tracking and disturbance compensation.

The task of limiting the frequency content for tracking feedforward $K_{ff,r}$ is shifted to the design of band-limited reference motion profile x_d . The industrial standard jerk-limited S-curve motion profile is used here [38, §5.6.2], maximum jerk and acceleration values of the S-curve profile are chosen such that the dominant effective excitation frequency of the reference trajectory is less than the required band limit $f_{c,r}$ of the tracking feedforward $K_{ff,r}$, which is described in [49]. Due to the inherent band limit of the selected reference signal, the additional low pass term of $K_{ff,r}$ can be neglected in the implementation to avoid unnecessary phase delay.

The modified disturbance feedforward $K_{ff,d}$ is realized as the exact inverse of the GP given in Eq. (11), followed by a third order lag term to represent the band limitation as in Eq. (17). The third order low pass filter is implemented in both forward and backward directions to remove the phase shift and to keep the disturbance feedforward synchronized with the tracking feedforward. Such a filtering strategy requires a preview of the reference trajectory x_d and its derivatives before the current time step, which is available in the industrial numerical control system by means of the look-ahead functionality [2]. Alternatively, this preview-based synchronization strategy can also be implemented by delaying the tracking feedforward accordingly.

D. Tracking Performance

To validate the steady-state tracking performance, which determines the surface finish quality of workpieces manufactured on a machine tool, constant velocity trajectories with $\dot{x}_d \in \{150, 175\}$ mm/s are chosen. Fig. 12 shows the steady-state tracking behavior with the corresponding feedforward controllers at $\dot{x}_d = 175$ mm/s. For a quantitative comparison, the tracking performance is evaluated with the mean absolute error (mae) and the maximum absolute error (max). The respective control effort is quantified by the standard deviation of input signals during this constant velocity experiment, summarized in Table III.

Compared to the baseline feedforward neglecting the mechanical compliance, the hybrid modeling approach with exact and modified robust model inverse cut the tracking error at both experiments by more than 61% in mae value and more than 36% in max value. Interestingly, the tracking behavior of the baseline feedforward is no longer offset free at $t \in [0.8, 1.4]$ s, resulting in a rather large average error. This is due to neglecting the axial kinematic errors which, especially at high velocities, leads to a velocity deviation between the drive motor and the load, see the slower varying part of Φ_{NL} in Fig. 10. To further illustrate the resulting vibration level, the tracking errors are detrended using a high pass filter with a cut-off frequency of 5 Hz. The hybrid modelling approach still reduces the detrended mae error by 21% with the exact inverse and by 26% with the robust modified inverse. Noticeably, the primary periodic disturbance due to the cyclical motion of the ball screw at $v = 175$ mm/s has a frequency of $f_{dist} = v/h = 35$ Hz with h the spindle lead. This is outside the bandwidth $f_b \approx 10$ Hz and can hardly be handled by the given feedback controller.

The control effort of the modified robust inverse is reduced by at least 47% compared to the exact inverse feedforward with comparable tracking error, since the modified robust inverse limits the feedforward gain of the high frequency content. This is as expected because the control input weight W_u is selected based on the measured FRFs from reference r to control signal u in the standard control loop, which consequently implies a comparable control effort to the standard feedforward, cf. Sec. IV-C.

In addition to the constant velocity phase, feed drives are particularly challenged in the transient phase during acceler-

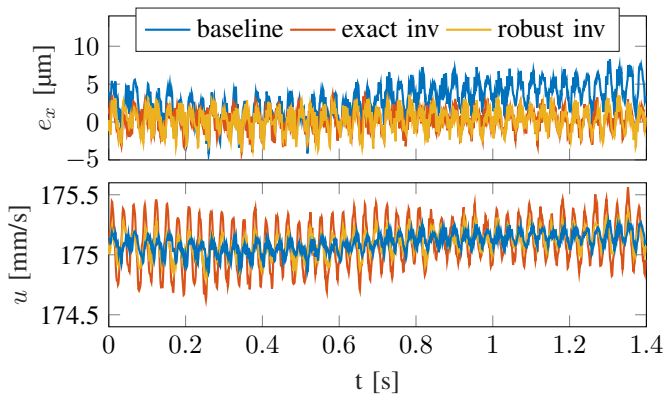


Fig. 12. Tracking error and control signal for constant velocity of 175 mm/s.

 TABLE III
 TRACKING PERFORMANCE AT CONSTANT VELOCITY.

	baseline	exact inv	robust inv
velocity: 150 mm/s			
mae(e_x) [μm]	2.79	1.06	1.03
max($ e_x $) [μm]	7.01	4.47	4.24
std(u) [mm/s]	0.07	0.17	0.09
velocity: 175 mm/s			
mae(e_x) [μm]	3.11	1.12	1.22
max($ e_x $) [μm]	8.28	4.08	3.89
std(u) [mm/s]	0.08	0.19	0.10

829 ation and deceleration, where the control performance deter-
 830 mines the part tolerance and the cycle time. Here the industrial
 831 standard jerk-limited S-curve motion profile is chosen [38,
 832 §5.6.2], and set to have a maximum velocity of 0.2 m/s, a
 833 maximum acceleration of 2 m/s² and a maximum jerk of
 834 10 m/s³ traveling along the entire axis range. The validation
 result is given in Fig. 13 and in Table IV.

 TABLE IV
 TRACKING PERFORMANCE OF THE RESPECTIVE CONTROLLERS.

	baseline	exact inv	robust inv
mae(e_x) [μm]	17.01	8.02	2.96
max($ e_x $) [μm]	90.71	52.39	16.24

835 In contrast to the baseline feedforward assuming rigid body
 836 dynamics of the control loop, it is observed that the exact
 837

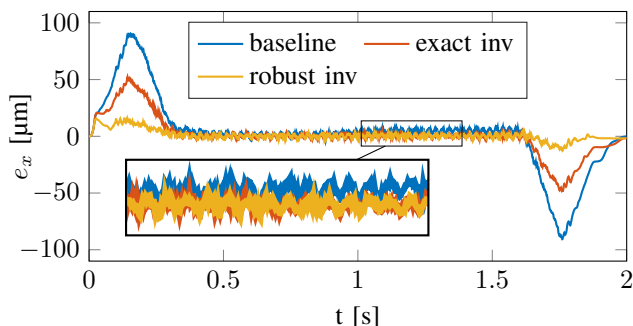


Fig. 13. Tracking error with jerk-limited S-curve motion profile.

inverse feedforward, which approximates the dynamics by a
 low-order model with only two parameters, reduces the track-
 ing error by 53% in mae value and by 42% in max value. This
 clearly illustrates the benefit of the selected analytical model
 structure dedicated to the feedforward design, as discussed
 in Sec. III-B. In addition, the modified robust feedforward,
 designed by the μ synthesis framework with optimized robust
 performance, cut the tracking error even further by more than
 82% in both metrics. It can be seen from Table II that the
 robust synthesis method sets a lower resonant frequency $\omega_{0,r}$
 and a higher damping ratio $D_{0,r}$ for tracking control than the
 identified model parameters, which leads to a more significant
 feedforward action in the low frequency range relevant for
 trajectory tracking and explains the reduction in tracking errors
 compared to the exact inverse.

Overall, the tracking performance with exact and robust
 model inversion is superior to the baseline feedforward in
 both steady and transient states, illustrating the benefit of the
 chosen hybrid structure for feedforward design. In addition,
 the proposed robust synthesis framework further optimizes the
 control performance compared to the nominal exact inverse,
 even with limited commissioning complexity.

E. Robustness Analysis

Apart from the tracking performance, the robustness of
 the proposed feedforward design approach is investigated
 experimentally. This is separated into robustness studies in
 the face of errors in the data-driven model and the analytical
 model.

The robustness test against underfitting and overfitting of the
 GP model is performed by setting the length scale parameter to
 $l_{1,\text{under}} = 0.005$ and $l_{1,\text{over}} = 0.0006$, respectively. The tracking
 result at $\dot{x}_d = 175$ mm/s is given in Fig. 14. Noticeably,
 despite the errors in GP model, the hybrid feedforward still
 ensures an offset-free tracking behavior, and reduces the over-
 all tracking error by more than 33% in mae value compared
 to the baseline. This is due to the correction of slower
 kinematics errors via the GP model, as discussed in Sec. V-D.
 Considering the resulting vibration level by detrending the
 tracking error, the underfitted GP model increases the error by
 52% and 11% for the exact and robust inverse, respectively,
 due to the incorrectly estimated periodic pattern of Φ_{NL} .
 As for the overfitting, the resulting vibration level remains
 similar to the baseline control for exact inverse (increased
 by 6%) and robust inverse (reduced by 5%). This illustrates
 the inherent robustness of our chosen model structure against
 overfitting: due to the low pass nature of the control loop with
 limited bandwidth, the overly high frequency input command
 resulting from the overfitted GP is no longer tracked by
 the underlying speed control loop, and therefore does not
 significantly increase the vibration level. Overall, the modified
 robust design achieves better worst case performance than the
 nominal design, and drastically reduces the control effort by
 55% and 82%, preventing potential input saturation, especially
 in the case of overfitting.

The robustness is also investigated with mismatched model
 parameters of the analytical model G_0 to simulate errors in the

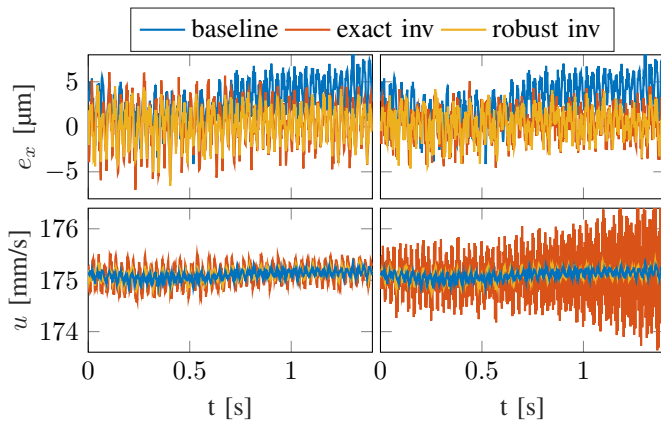


Fig. 14. Robustness to wrong GP model (left: underfitting, right: overfitting).

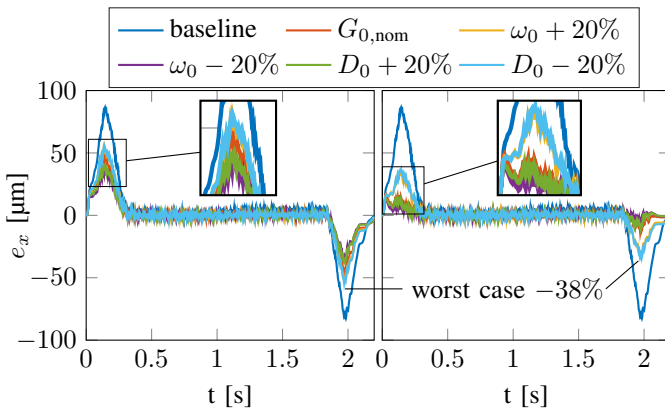


Fig. 15. Robustness to wrong analytical model (left: exact inverse, right: modified robust inverse).

identification or varying plant dynamics. The nominal model parameters ω_0 and D_0 are varied by $\pm 20\%$ respectively for the inverse feedforward control with S-curve motion profile. For the robust feedforward synthesis, the uncertainty weight W defined in Eq. (14) must be chosen appropriately to adapt the uncertainty set to the deliberately varied model parameters, while the other weights of the robust synthesis problem remain the same. The result in Fig. 15 shows that, even the nominal exact inverse feedforward with significant model errors still achieves a performance improvement of at least 35% compared to the baseline feedforward with rigid body assumption. Furthermore, the presented robust synthesis method improves the worst case performance by 38% in comparison to the nominal feedforward.

Overall, this experimental robustness analysis illustrates the excellent resilience to errors in the model parameters of the inverse feedforward design with the chosen model structure, and the significantly increased robustness of the presented robust inversion solution as opposed to the exact inversion.

VI. CONCLUSION

We presented an inversion-based feedforward design approach for the feed drive control system based on hybrid modeling. The hybrid model, developed with a particular focus on

its use for real-time feedforward compensation, combines a flat analytical model of linear dynamics and a GP model of output nonlinearities. Besides the exact model inversion solution, the main design contribution is a robust inversion-based feedforward control that explicitly accounts for model uncertainties. The robust synthesis scheme is adopted to optimize the robust performance of the feedforward control under uncertainties. To increase the practical applicability, the synthesis problem of feedforward controllers is formulated in a signal-based manner, and the commissioning complexity of feedforward gains is reduced to the selection of two hyperparameters. Extensive experimental results on an industrial milling machine illustrate the real-time capability and significant performance improvement of the robust feedforward control with hybrid model. Furthermore, the excellent robustness to errors in the analytical model and the data-driven model of this feedforward synthesis framework is demonstrated experimentally.

Future work includes representing the disturbance term by the frequency domain GP model, as in [50], which may provide a more accurate, higher fidelity uncertainty quantification and reduce conservatism. The practical challenge is that the disturbance transfer function tends to have very small magnitudes, requiring more appropriate treatment of numerical issues.

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Haijia Xu received the M.Sc. degree in engineering cybernetics from the University of Stuttgart, Stuttgart, Germany, in 2022. He is currently pursuing the Ph.D. degree with the Institute for Control Engineering of Machine Tools and Manufacturing Units, University of Stuttgart, Stuttgart, Germany. His current research interests include hybrid modeling and robust control methods applied to mechatronic systems.

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Christoph Hinze received the M.Sc. degree in engineering cybernetics and the Ph.D. (Dr.-Ing.) degree in mechatronics from the University of Stuttgart, Stuttgart, Germany, in 2017 and 2023, respectively.

Since 2023 he has been leading a junior research group that is investigating hybrid analytical and data-driven modelling approaches for path planning and control of manufacturing systems. His research interests lie in modelling the dynamics of mechatronic systems and practical applications to industrial motion control, path planning and feedforward control.

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Andrea Iannelli (Member, IEEE) is an Assistant Professor in the Institute for Systems Theory and Automatic Control at the University of Stuttgart (Germany). He completed his B.Sc. and M.Sc. degrees in Aerospace Engineering at the University of Pisa (Italy) and received his PhD from the University of Bristol (United Kingdom) on robust control and dynamical systems theory. He was a postdoctoral researcher in the Automatic Control Laboratory at ETH Zürich (Switzerland). His main research interests are at the intersection of control

theory, optimization, and learning, with a particular focus on robust and adaptive optimization-based control, uncertainty quantification, and sequential decision-making problems. He serves the community as Associated Editor for the International Journal of Robust and Nonlinear Control and as IPC member of international conferences in the areas of control, optimization, and learning.

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Alexander Verl (Member, IEEE) received the Dipl.-Ing. degree in electrical engineering from Friedrich-Alexander-University Erlangen-Nürnberg, Erlangen, Germany, in 1991, and the Dr.-Ing. degree in control engineering at the DLR Institute of Robotics and Mechatronics, Oberpfaffenhofen, Germany, in 1997. He is a Full Professor since 2005, and the head of the Institute for Control Engineering of Machine Tools and Manufacturing Units (ISW) at the University of Stuttgart, Stuttgart, Germany. Prof. Verl received the Honorary Doctorate (Dr. h.c.) from the Polytechnic

University of Timisoara, Romania, in 2009, the Invention and Entrepreneurship Award from the IEEE Robotics and Automation Society, the International Federation of Robotics, in 2010, the Dr. h.c. from the Technical University of Cluj-Napoca, Romania, in 2012, the Honorary Professor from the University of Auckland, New Zealand, in 2012, the Julius von Haast Fellowship Award from the Royal Society of New Zealand in 2012, and the Diesel Gold Medal from the German Institute of Inventions in 2014.