1 2

COMPUTATION OF BIFURCATION MARGINS BASED ON ROBUST CONTROL CONCEPTS

3

ANDREA IANNELLI^{*}, MARK LOWENBERG [†], AND ANDRÉS MARCOS[†]

Abstract. This article proposes a framework which allows the study of stability robustness of 4 equilibria of a nonlinear system in the face of parametric uncertainties from the point of view of 5 6 bifurcation theory. In this context, a branch of equilibria is stable if bifurcations (i.e. qualitative changes of the steady-state solutions) do not occur as one or more bifurcation parameters are varied. 8 The work focuses specifically on Hopf bifurcations, where a stable branch of equilibria meets a 9 branch of periodic solutions. It is of practical interest to evaluate how the presence of uncertain parameters in the system alters the result of analyses performed with respect to a nominal vector 10 field. Note that in this article bifurcation parameters have a different meaning than uncertain 11 12 parameters. To answer the question, the concept of robust bifurcation margins is proposed based 13 on the idea of describing the uncertain system in a Linear Fractional Transformation fashion. The 14 robust bifurcation margins can be interpreted as nonlinear analogs of the structural singular value, or μ , which provides robust stability margins for linear time invariant systems. Their computation 15 is formulated as a nonlinear program aided by a continuation-based multi-start strategy to mitigate 1617 the issue of local minima. Application of the framework is demonstrated on two case studies from 18 the power system and aerospace literature.

19 **Key words.** Bifurcations, numerical continuation, robust control theory, robust stability

20 AMS subject classifications. 34C23, 34H20, 37G15, 93D09

1. Introduction. Bifurcation analysis studies qualitative changes in the re-21 sponse of a nonlinear system (e.g. number and type of steady-state solutions) when 22 one or more parameters on which the dynamics depend are continuously varied 23 [27, 21]. This is usually accomplished by selecting a few *bifurcation parameters*, typ-24 ically equal in number to the codimension of the studied bifurcation, based on their 25importance for the system. This analysis approach is of recognized importance since 26 it allows complex dynamic behaviours to be characterized and an understanding of 27the system to be gained. However, it does not provide indications on the robustness 28of the results to uncertainties in the models. Let us consider for example the presence 29of *uncertain parameters* allowed to vary within a prescribed range. These parameters 30 31 reflect the fact that uncertainty is ubiquitous in engineering systems and at any stage of analysis (from preliminary to detailed). Unlike the bifurcation parameters, in principle they are not restricted in number (and are allowed to vary simultaneously) and 33 their influence on the dynamics may not be known a priori. It is then important to 34estimate their effect, and in particular whether bifurcation points can move closer to 35 36 operating points deemed safe on the basis of analyses applied to the nominal system. The study of robustness within a dynamical systems perspective can be attempted 37 by adopting singularity theory techniques (e.g. Lyapunov-Schmidt reduction) [18], as 38 shown by recently published research [20, 8]. The central idea is to perform a reduc-39 tion of the original dynamics to a lower dimension map, whose singularities represent 40 41 transitions between qualitatively different bifurcation diagrams. Even though it is in principle possible to track these singularities without computing explicitly the reduc-42

43 tion [8], the application of these techniques to systems with a moderately complex

44 mathematical description and with generic number of uncertainties is not straight-

^{*}Department of Information Technology and Electrical Engineering, Swiss Federal Institute of Technology (ETH), Zürich 8092, Switzerland (iannelli@control.ee.ethz.ch).

[†]Department of Aerospace Engineering, University of Bristol, BS8 1TR, United Kingdom (m.lowenberg/andres.marcos@bristol.ac.uk).

forward and has not been presented in the literature yet. Moreover, this approach does not directly provide information on the *distance* from a given (nominally stable) operating point to the closest bifurcation, that is, a margin to the bifurcation. Another approach which considers the effect of uncertainties by focussing on a reduced dimensional dynamics, namely the one on the centre manifold, is that proposed in [38]. The main difficulty resides here in the definition of appropriate initial conditions allowing a projection of the long term dynamics on the centre manifold which accurately incorporates the effect of uncertainties [37].

This article proposes a framework which provides a quantitative measure of the distance between branches of stable equilibria and of periodic oscillations in the uncer-54tainty space. In other words, the onset of a Hopf bifurcation in the face of worst-case combinations of the uncertainty is formalised by means of a robust bifurcation margin. 56 Previous works in the literature looked at the problem of computing perturbations to bifurcations. For example, in [12] an extension to multidimensional parameter 58 spaces of standard methods for codimension-1 bifurcations is proposed. The problem of determining locally closest bifurcations is solved by introducing a normal vector 60 61 to hypersurfaces of bifurcation points, and makes use of both direct and iterative methods. While the latter is limited to static bifurcations (i.e., saddle node, tran-62 scritical, and pitchfork), the former is in principle applicable also to the Hopf case. 63 The direct method consists of solving the full set of equations defining the bifurcation 64 (plus additional equations to close the problem) and, as pointed out by the authors 65 of [12], it may be too onerous from a computational point of view. This approach 66 67 was applied in [32] to the analysis of static bifurcations in flexible satellites, making a number of simplifying assumptions, e.g., no dependence of the equilibrium on the un-68 certainties and the system having Hamiltonian dynamics. A closely related approach, 69 which according to their authors generalizes the method from [12], is discussed in [6]. 70 The work considers saddle-node bifurcations only, and the computation of the small-71est perturbation to bifurcation is done by applying the generalized reduced-gradient 7273 method. In essence, this consists of a nonlinear optimization strategy making use of corrector and predictor steps and solving the system of equations defining the bifur-74 cation. However, the issue of local minima is not addressed and the same objection 75regarding the total dimension of the problem is envisaged for the Hopf bifurcation 76case (not discussed in that work). The idea of using vectors normal to a manifold of 77 bifurcation points is also present in [16, 34] and other works from the same group, 78 79 where the design of robustly stable and feasible processes is pursued.

The problem is studied in this article from the point of view of Linear Fractional 80 Transformation (LFT) models and structured singular value (μ) analysis from ro-81 bust control theory [48]. These tools are well established for the analysis of linear 82 uncertain systems, and provide an analytical answer to stability and performance 83 problems. Even though a direct application to the nonlinear context is precluded 84 by their inherently linear formulation, an extension is proposed here for computing 85 robust bifurcation margins. The core idea is to build an LFT model of the Jacobian 86 of the uncertain vector field (which will generically depend on the states of the system 87 88 and on the uncertainties) and to formulate the computation of the closest Hopf bifurcation as the worst-case perturbation matrix for which the LFT becomes singular. 89 90 This bears similarities to the problem solved by μ , but significant differences hold as commented in the paper. The determination of the margins is posed as a nonlinear smooth optimization problem, which can be solved with off-the-shelf algorithms. The 92 program also allows the type of Hopf bifurcation (subcritical or supercritical) to be 93 specified by constraining the sign of the first Lyapunov coefficient. Since the opti-94

95 mization problem is nonlinear, the issue of local minima is discussed and different 96 strategies are proposed to mitigate it. These include a multi-start strategy based on 97 the construction of a manifold of Hopf points connected to a given solution obtained 98 by the optimizer. The main advantages of the proposed approach, whose formulation 99 is detailed in section 3, include: low dimension and computational cost of the solved 90 problem; improved confidence on the accuracy of the results in terms of global validity

101 of the optimum; possibility to apply the wealth of analysis strategies available with μ 102 (e.g., sensitivity analysis, frequency interpretation of the results).

In section 4 the use of this framework to study nonlinear stability problems arising 103 in power system and aerospace applications is investigated by considering two case 104studies from the literature. First, the sensitivity to a set of physical parameters of the 105Hopf bifurcation encountered in a power load system with voltage regulator and dy-106 namic load model is considered in section 4.1. It is shown that the application of the 107 robust bifurcation margin allows on one hand to retrieve the same findings reported 108 in [13] (which considered a first-order approximation of the sensitivity), and on the 109other to investigate more sophisticated types of sensitivity analyses where coupling 110 111 among uncertain parameters are also accounted for.

112 Then, an aeroelastic flutter case study is analyzed in section 4.2. Flutter is a selfexcited instability in which aerodynamic forces on a flexible body couple with its 113natural vibration modes producing oscillatory motion. In the presence of nonlineari-114 ties, the system typically exhibits loss of stability of the equilibrium in the form of a 115Hopf bifurcation with ensuing Limit Cycle Oscillations (LCO). Results show a good 116 117 match with prior studies that considered linear robust analyses [25], and highlight the unique capability of this framework to allow the type of Hopf bifurcation (subcritical 118 or supercritical) of which robustness is studied to be chosen in the analysis. 119

120 Bifurcation analysis has been extensively applied to both application fields [41, 11],

but the effect of uncertainties has received far less attention. The results in section 4 show that the proposed framework can be a valuable tool for analyzing robustness in the nonlinear context and a more in depth application to these challenging problems is a future research direction.

125 Preliminary results of this work were presented in [24].

<u>Notation</u>: [x; y] denotes vertical concatenation of two vectors $x \in \mathbb{R}^n$ and $y \in \mathbb{R}^m$. 126 $|\mathbb{I}|$ indicates cardinality of a set $\mathbb{I}, \, \bar{\sigma}(P)$ is the maximum singular value of a matrix 127 $P \in \mathbb{R}^{n \times n}$, \bar{r} is the complex conjugate of $r \in \mathbb{C}^n$ and $\langle r, q \rangle = \bar{r}^T q$ is the scalar product 128 129 between complex vectors $r, q \in \mathbb{C}^n$. Where evident from the context, subscripts of vectors and matrices are used to specify their elements (e.g., x_3 is the third element of 130 $x \in \mathbb{R}^n$; the symbol is used for solutions of an optimization; the symbol is used for 131 uncertain quantities; diag (\cdot) indicates a block diagonal matrix made up of elements 132133 in ·.

2. Background. This section provides an overview on the techniques and tools employed in the work. The first subsection presents the theoretical background of bifurcation (2.1.1) and numerical continuation (2.1.2). This is followed by a short introduction to the robust control concepts of LFT models (2.2.1) and μ analysis (2.2.2).

139 **2.1.** Nonlinear dynamics approaches.

140 2.1.1. Bifurcation theory. Consider an autonomous nonlinear system of the141 form

142 (2.1)
$$\dot{x} = f(x, p)$$

where $x \in \mathbb{R}^{n_x}$ and $p \in \mathbb{R}^{n_p}$ are respectively the vectors of states and bifurcation parameters, and $f : \mathbb{R}^{n_x} \times \mathbb{R}^{n_p} \to \mathbb{R}^{n_x}$ is the vector field. In this work f is assumed to gather smooth nonlinear functions $(f \in \mathcal{C}^{\infty})$. Therefore, the Jacobian matrix of the vector field $\nabla_x f : \mathbb{R}^{n_x} \times \mathbb{R}^{n_p} \to \mathbb{R}^{n_x \times n_x}$, denoted here by J, is always defined.

The vector x_0 is called a fixed point or equilibrium of (2.1) corresponding to p_0 147if $f(x_0, p_0) = 0$. Let us denote with n_0 the number of eigenvalues of $J(x_0, p_0)$ with 148 zero real parts, respectively. Then x_0 is called a hyperbolic fixed point if $n_0 = 0$, 149otherwise it is called nonhyperbolic. Bifurcations of fixed points are concerned with 150the loss of hyperbolicity of the equilibrium as p is varied. Two scenarios can take 151place: static bifurcations and dynamic bifurcations [27, 21]. The former arise when 152J is singular at an equilibrium, i.e., it has a zero eigenvalue. The common feature 153of static bifurcations is that branches of fixed points meet at the bifurcation point. 154In the case of dynamic bifurcations, branches of fixed points and periodic solutions 155meet. This case, also referred to as Hopf bifurcation, is the focus of this work and is 156formally described by the following theorem. 157

158 THEOREM 2.1 ([21] Hopf bifurcation theorem). Suppose that the system $\dot{x} = f(x,p), x \in \mathbb{R}^{n_x}$ and $p \in \mathbb{R}$ has an equilibrium (x_H, p_H) at which the following 160 properties are satisfied.

161 1. $J(x_H, p_H)$ has a simple pair of pure imaginary eigenvalues and no other ei-162 genvalues with zero real parts. This implies, for the implicit function theorem, 163 that there is a smooth curve of equilibria (x(p), p) with $x(p_H) = x_H$. The ei-164 genvalues $\nu(p)$, $\bar{\nu}(p)$ of J(x(p)), with $\nu(p_H) = i\omega_H$, vary smoothly with p. 165 2. It holds

166 (2.2)
$$\frac{d}{dp} \left(\operatorname{Re} \nu(p) \right) \Big|_{p=p_H} = l_0 \neq 0.$$

167 Then there is a unique three-dimensional center manifold passing through (x_H, p_H) in 168 $\mathbb{R}^{n_x} \times \mathbb{R}$ and a smooth system of coordinates for which the Taylor expansion of degree 169 3 on the center manifold is given in polar coordinates (ρ, θ) by

170 (2.3)
$$\dot{\rho} = (l_0 p + l_1 \rho^2) \rho, \\ \dot{\theta} = \omega + l_2 p + l_3 \rho^2,$$

171 where l_0 , l_1 , l_2 , and l_3 are real coefficients defining the manifold. If $l_1 \neq 0$, there is a

surface of periodic solutions in the center manifold which has quadratic tangency with the eigenspace of $\nu(p)$, $\bar{\nu}(p)$. If $l_1 < 0$, then these periodic solutions are stable limit

174 cycles, while if $l_1 > 0$, the periodic solutions are repelling.

Note first that the theorem is typically stated considering a scalar p since the Hopf 175bifurcation is codimension-1. Condition 1 of Th. 2.1 requires that the Jacobian of the 176 vector field has a pair of purely imaginary eigenvalues (and no other eigenvalues on the 177 178imaginary axis). Condition 2, also known as the transversality condition, prescribes that these eigenvalues are not stationary with respect to p at the bifurcation. A 179fundamental parameter determining the dynamic behaviour in the neighborhood of a 180 Hopf point is l_1 , also called the first Lyapunov coefficient. Its value determines whether 181the Hopf bifurcation is *subcritical* or *supercritical*, and its analytical expression is given 182by [27] 183

184 (2.4)
$$l_1 = \frac{1}{2\omega_H} \operatorname{Re}\langle r, C(q, q, \bar{q}) - 2B(q, A^{-1}B(q, \bar{q})) + B(\bar{q}, (2i\omega_H I_n - A)^{-1}B(q, q)) \rangle.$$

185 Here the complex vectors $r, q \in \mathbb{C}^{n_x}$ satisfy

186 (2.5)
$$Jq = i\omega_H q, \qquad J^T r = -i\omega_H r, \qquad \langle r, q \rangle = 1.$$

187 The functions $B : \mathbb{R}^{n_x} \times \mathbb{R}^{n_x} \to \mathbb{R}^{n_x}$ and $C : \mathbb{R}^{n_x} \times \mathbb{R}^{n_x} \to \mathbb{R}^{n_x}$ are the tensors

of second and third order derivatives evaluated at x_H , respectively. For example, for vectors ξ , ς , $\chi \in \mathbb{R}^{n_x}$, $B(\xi, \varsigma)$ and $C(\xi, \varsigma, \chi)$ are in \mathbb{R}^{n_x} with components

$$B_i(\xi,\varsigma) = \sum_{j,k=1}^{n_x} \left. \frac{\partial^2 f_i(x,p)}{\partial x_j x_k} \right|_{x=x_H,p=p_H} \xi_j \varsigma_k, \quad i = 1, 2, ..., n_x,$$
$$C_i(\xi,\varsigma,\chi) = \sum_{j,k,l=1}^{n_x} \left. \frac{\partial^3 f_i(x,p)}{\partial x_j x_k x_l} \right|_{x=x_H,p=p_H} \xi_j \varsigma_k \chi_l, \quad i = 1, 2, ..., n_x.$$

2.1.2. Numerical continuation. The computational tool of bifurcation analy-191 sis is numerical continuation, providing path following algorithms allowing implicitly 192defined manifolds [19] to be computed. These schemes are based on the implicit func-193194 tion theorem (IFT) [45], which guarantees, under the condition that J is non-singular at an initial point (x_0, p_0) , that there exist neighbourhoods X of x_0 and P of p_0 and 195a function $g: P \to X$ such that f(x, p) = 0 has the unique solution x = g(p) in X. 196Examples of numerical techniques to compute the implicit manifold g are Newton-197 Raphson, arclength, and pseudo-arclength continuation [19], efficiently implemented 198 in freely available software, e.g., AUTO [14], and COCO [10]. 199

A general continuation problem, so called *extended*, can be formulated as follows [9, 10]

202 (2.7)
$$F(u,\lambda) := \begin{pmatrix} \Phi(u) \\ \Psi(u) \end{pmatrix} - \begin{pmatrix} 0 \\ \lambda \end{pmatrix} = 0,$$
$$\Phi : \mathbb{R}^{n_u} \to \mathbb{R}^m, \qquad \Psi : \mathbb{R}^{n_u} \to \mathbb{R}^{n_\lambda},$$

where Φ defines the zero problem in the vector u of continuation variables, Ψ denotes 203a family of monitor functions and λ is a vector of continuation parameters. It is 204 straightforward to see that the goal of tracking equilibria of the vector field f can 205be pursued by solving the zero problem only with $\Phi = f$, and u = [x; p]. However, 206207 the extended continuation problem in (2.7) allows for a greater variety of problems to be solved, as the related concept of *restricted* continuation problem shows. Let 208 $\mathbb{I} \subseteq \{1, ..., n_{\lambda}\}$ be an index set and $\overline{\mathbb{I}}$ its complement in $\{1, ..., n_{\lambda}\}$. Let $\lambda_{\mathbb{I}} = \{\lambda_i | i \in \mathbb{I}\}$ 209and consider the restriction $F(u,\lambda)|_{\lambda_{I}=\lambda_{I}^{*}}$ satisfying the IFT at some point $(u^{*},\lambda^{*}=$ 210 $\Psi(u^*)$). Then $F(u,\lambda)|_{\lambda_{\mathbb{I}}=\lambda_{\mathbb{I}}^*}$ defines a continuation problem for a *d*-manifold with 211 $d = n_u - (m + |\mathbb{I}|)$. $\lambda_{\overline{\mathbb{I}}}$ and $\lambda_{\mathbb{I}}$ are called the set of active and inactive continuation 212parameters respectively, since the former changes during continuation, while the latter 213 remain constant. Analogously, equations corresponding to $\lambda_{\mathbb{T}}$ are inactive constraints, 214while equations corresponding to $\lambda_{\mathbb{I}}$ are active constraints, because they impose an 215additional condition on the solutions to the set of zero problems. The formulation 216(2.7) is implemented in the software COCO, which is used for all the continuation 217218 analyses performed in this work.

219 **2.2. Robust control theory.**

220 **2.2.1. The Linear Fractional Transformation paradigm.** Linear Fractional 221 Transformation (LFT) is an instrumental tool in robust control theory for analysis and control of uncertain systems [48]. For the sake of clarity, first an intuition of the reasoning behind LFT is given, followed by a more formal definition.

The classic interpretation of an LFT is in terms of input to output relationship of 224 a feedback interconnection. Let us consider a linear time invariant (LTI) system with 225transfer matrix (i.e. matrix of transfer functions) $M_{22} \in \mathbb{C}^{p_2 \times q_2}$, input v and output 226y. The system M_{22} is assumed to be exactly known, and thus is also termed *nominal*. 227 If the model has uncertainties (which will be better characterized later), these can be 228 modelled with an operator $\Delta_u \in \mathbb{C}^{q_1 \times p_1}$ with input z and output w. The effect of Δ_u 229on M_{22} can then be described by introducing the transfer matrices M_{11} , M_{12} and M_{21} . 230 For example, in the case of parametric uncertainties, these will be simply static (gain) 231matrices, while for the case of unmodelled dynamics these could also have dynamic 232 233 terms (e.g. low pass filters). The key point is that, by choosing these matrices, the analyst can describe with a certain flexibility how the perturbation affects the nominal 234system. Given this setting, Figure 1 shows the standard representation of LFT.

FIG. 1. Standard feedback representation of an LFT.

The central idea is thus to represent the uncertain system as a feedback of known components (the transfer matrices M_{ij}) with uncertain (the operator Δ_u) ones. In practice, this is done by pulling out of the system the unknown parts, so that the problem appears as a nominal system subject to an artificial feedback. Available toolboxes [28] allow this operation to be efficiently performed and provide the analyst, given a description of how the uncertainties affect the system, with the matrices M_{ij} . In order to formally define an LFT, let us denote by $M \in \mathbb{C}^{(p_1+p_2)\times(q_1+q_2)}$ the

243 partitioned transfer matrix (also termed *coefficient matrix*)

244 (2.8)
$$M = \begin{bmatrix} M_{11} & M_{12} \\ M_{21} & M_{22} \end{bmatrix},$$

and let $\Delta_u \in \mathbb{C}^{q_1 \times p_1}$ the uncertain operator. The LFT of M with respect to Δ_u is defined as the map $\mathcal{F} : \mathbb{C}^{q_1 \times p_1} \longrightarrow \mathbb{C}^{p_2 \times q_2}$

247 (2.9)
$$\mathcal{F}(M, \Delta_u) = M_{22} + M_{21}\Delta_u (I - M_{11}\Delta_u)^{-1} M_{12}.$$

With reference to Fig. 1, $\mathcal{F}(M, \Delta_u)$ compactly defines the uncertain transfer matrix 248 from input v to output y of the nominal system M_{22} when this is subject to Δ_u . 249Indeed, for $\Delta_u = 0$ (no uncertainties in the model) it holds $\mathcal{F}(M, \Delta_u) = M_{22}$. It is 250also important to observe that M_{11} is, within this input to output framework, the 251transfer matrix seen by the perturbation block Δ_u . A crucial feature apparent in (2.9) 252253 is that the LFT is well posed if and only if the inverse of $(I - M_{11}\Delta_u)$ exists. Otherwise, $\mathcal{F}(M, \Delta_u)$ is said to be singular. Singularity of the LFT is typically associated with the 254loss of stability of the underlying uncertain system, and thus finding the uncertain 255perturbations for which this happens is typically the objective of robust stability 256257analysis (details on this will be provided in Sec. 2.2.2).

235

In robust control, Δ_u typically gathers parametric and dynamic uncertainties and can be represented as

260 (2.10)
$$\Delta_u = \operatorname{diag}(\delta_i I_{d_i}, \delta_j I_{d_j}, \Delta_{D_k}), i = 1, ..., n_R, \quad j = n_R + 1, ..., n_R + n_C, \quad k = 1, ..., n_D,$$

where the uncertainties associated with n_R real scalars δ_i , n_C complex scalars δ_j , 261 and n_D unstructured (or full) complex blocks Δ_{D_k} are listed in diagonal format. 262 The identity matrices of dimension d_i and d_j take into account the fact that scalar 263 uncertainties might be repeated in Δ_u when the LFT of the system is built up. For 264example, if a matrix has the parameter δ_i on three different rows, in order to cast 265266 it in the form of an LFT (2.9) it will be necessary to have $d_i=3$ [28]. Typically the uncertain parameters are normalized by scaling of M such that $\Delta_u = 0$ coincides with 267the nominal system (i.e., uncertain parameters at their nominal values) and $\bar{\sigma}(\Delta_u) \leq 1$ 268 when uncertainties take values in the allowed interval. The set in (2.10) is generally 269referred to as *structured* because of the block diagonal structure. This feature, enabled 270by the LFT modeling paradigm, is known to provide less conservative results in the 271272analysis of uncertain systems with respect to unstructured representations (used, for example, in the celebrated small gain theorem [48]). 273

This work leverages the LFT framework for analysis of nonlinear systems. The interpretation given previously, while providing insights into this paradigm, cannot be thus readily used since it requires transfer matrices. For this reason, an alternative viewpoint on LFT is proposed.

Let us start by considering the state-space (SS) representation $(\mathcal{A}, \mathcal{B}, \mathcal{C}, \mathcal{D})$ of the nominal LTI system with transfer matrix M_{22}

280 (2.11a) $\begin{cases} \dot{x} = \mathcal{A}x + \mathcal{B}v, \\ y = \mathcal{C}x + \mathcal{D}v, \end{cases}$

284 (2.11b)
$$M_{22}(s) = \mathcal{D} + \mathcal{C}(sI_{n_x} - \mathcal{A})^{-1}\mathcal{B},$$

283 where s is the Laplace variable. Define now

284 (2.12)
$$M_{\nu} = \begin{bmatrix} \mathcal{A} & \mathcal{B} \\ \mathcal{C} & \mathcal{D} \end{bmatrix}, \quad \Delta_{\nu} = \frac{1}{s} I_{n_x}.$$

285 It can then be shown that $\mathcal{F}(M_{\nu}, \Delta_{\nu}) = M_{22}$. This follows directly from

286 (2.13)
$$\mathcal{F}(M_{\nu}, \Delta_{\nu}) = \mathcal{D} + \mathcal{C}\frac{1}{s}I_{n_{x}}(I_{n_{x}} - \frac{1}{s}\mathcal{A})^{-1}\mathcal{B} = \mathcal{D} + \mathcal{C}(sI_{n_{x}} - \mathcal{A})^{-1}\mathcal{B} = M_{22}(s),$$

where the diagonal structure of Δ_{ν} and the fact that $\frac{1}{s} \neq 0$ have been exploited. This result shows that the LFTs generalize the realization of transfer matrices into state-space (SS) representations to the case of rational multivariate matrices. For this reason, the LFT paradigm can also be regarded as a realization technique [28].

This interpretation also highlights a paramount aspect for the present work. The poles of (2.11) are typically found via eigenvalue analysis of \mathcal{A} . Equivalently, the system has a given pole ν if $(\nu I_{n_x} - \mathcal{A})^{-1}$ is singular. Note that this latter condition can be formulated as the singularity of the LFT $\mathcal{F}(M_{\nu}, \Delta_{\nu})$ by replacing $s = \nu$. In particular, the LTI (2.11) has a purely imaginary eigenvalue (i.e. it is neutrally stable) if there exists $\omega > 0$ for which $\mathcal{F}(M_{\nu}, \Delta_{\nu})$ is singular with $s = i\omega$.

Let us consider now the case when the LTI system (2.11) is subject to uncertainties. The problem can be described with the LFT formalism considering two blocks for the *uncertain* operator, namely Δ_u containing the structured perturbations, and Δ_{ν} . The coefficient matrix M is partitioned correspondingly

301 (2.14)
$$M = \begin{bmatrix} \mathcal{A} & \mathcal{A}_{12} & \mathcal{B} \\ \mathcal{A}_{21} & \mathcal{A}_{22} & \mathcal{B}_1 \\ \mathcal{C} & \mathcal{C}_1 & \mathcal{D} \end{bmatrix}, \quad \Delta = \operatorname{diag}(\Delta_{\nu}, \Delta_u)$$

302 A pictorial representation of the LFT $\mathcal{F}(M, \Delta)$ defined by the operators in (2.14) is

303 given in Figure 2.



FIG. 2. LFT of an uncertain state-space model.

The difference between the representations in Figure 1 and Figure 2, both describing an uncertain system, is that in the former the system is described via its transfer matrices, while in the latter a state-space representation is used. One can switch from the first to the second representation by exploiting the fact that $\mathcal{F}(M_{\nu}, \Delta_{\nu}) = M_{22}$ (which was proved above).

The consequence of this change of representation is that the new block Δ_{ν} appears. 309 Correspondingly, the coefficient matrix M (2.14) now features the matrix M_{ν} (2.12) 310 plus other matrices describing the effect of the uncertainties on the state-matrices. 311 Note indeed that the transfer matrices M_{11} , M_{12} and M_{21} will also be expressed here 312 313 with their SS representation. Let us assume now that (2.11) is nominally stable (i.e. 314 \mathcal{A} has all the eigenvalues in the left half-plane). Then the uncertain LTI system has a purely imaginary eigenvalue if there exist $\omega > 0$ (with $s = i\omega$) and a combination 315of the uncertainties in Δ_u for which $\mathcal{F}(M, \Delta)$ (2.14) is singular. 316

The advantage of this representation, which is key for the present work, is that LFTs can be constructed even for systems which do not have transfer matrices, if an *appropriate* state-space description is available. Sec. 3.1 will be devoted to showing which crucial steps can be taken in order to apply this rationale to the prototype of vector field introduced in (2.1).

Note finally that a useful property when dealing with LFTs featured by distinct Δ -blocks is that interconnections of LFTs can be rewritten as one single LFT. This is only a numerical aspect relative to the construction of LFT models, but it greatly helps to separate modeling-specific details of the system under consideration and to ease the algebraic manipulations. By virtue of this, it holds for the LFT defined in (2.14)

328 (2.15)
$$\mathcal{F}(M,\Delta) = \mathcal{F}(\mathcal{F}(M,\Delta_{\nu}),\Delta_{u}).$$

2.2.2. μ analysis. The μ analysis technique leverages the key features of LFT modeling reviewed in the previous section to address the robust stability analysis of LTI systems in the face of uncertainties. The structured singular value is a matrix function denoted by $\mu_{\Delta}(M)$ and several equivalent definitions are available in the literature [48, 15, 35]. A definition which encompasses the aspects relevant to this work is

335 (2.16)
$$\mu_{\Delta}(M) = \left(\min_{\Delta}(\kappa: \mathcal{F}(\mathcal{F}(M, \Delta_{\nu}), \Delta_{u}) \text{ is singular, } \bar{\sigma}(\Delta_{u}) \le \kappa\right)^{-1}$$

where κ is a real positive scalar, and $\mu_{\Delta}(M) = 0$ if the minimization problem has no solution.

Based on the point of view of LFT as realization technique, an interpretation of 338 the μ analysis technique is as *worst-case* eigenvalue analysis for uncertain systems. 339 Let us focus on the operator Δ of the LFT $\mathcal{F}(M, \Delta)$ defined in (2.14). The block 340 Δ_{ν} does not represent a true uncertainty of the system, and its meaning is that the 341 singularity of the LFT is checked against all the possible eigenvalues on the imaginary 342 axis. For the sake of understanding, one can think of realizing this block by considering 343 a set of frequencies ω and evaluating Δ_{ν} at $\nu = i\omega$. By doing this, $\Delta = \Delta_u$ and the 344 problem defined in (2.16) consists of finding the perturbation matrix with the smallest 345 maximum singular value (also termed worst-case matrix) such that the uncertain 346 system has a pair of purely imaginary eigenvalues $\pm i\omega$. Therefore, $\mu_{\Delta}(M)$ provides a 347 robust stability (RS) test for an uncertain linear system. Specifically, if $\mu_{\Delta}(M) \geq 1$ a 348 candidate (i.e., within the allowed range of the uncertainty set) perturbation matrix 349 exists that violates the well-posedness of $\mathcal{F}(M, \Delta)$. In essence, the uncertain state-350 matrix has the eigenvalues $s = \pm i\omega$ for a certain combination of the uncertainties 351 in the allowed range. On the contrary, if $\mu_{\Delta}(M) < 1$ then there is no perturbation 352 matrix inside the set Δ such that the $\mathcal{F}(M, \Delta)$ is ill-posed and thus the system is 353 robust stable within the range of uncertainties considered. 354

In the most established algorithms [2], μ is evaluated on a discretized frequency 355 range. That is, the Δ_{ν} block is realized as discussed before on a pre-selected grid 356 of frequencies, and the corresponding set of matrices $M(i\omega)$ (the dependence on the 358 frequency is now stressed) is computed. Subsequently, $\mu_{\Delta}(M(i\omega))$ is computed and a frequency-domain representation of the results is obtained. This is done in order to avoid the need to solve the optimization problem (2.16) on a continuous range of 360 frequency, which proves computationally challenging. An exception to this common 361 practice worth mentioning is represented by recently developed Hamiltonian-based 362 algorithms (i.e. SMART library [39] and Robust Control Toolbox from MATLAB 363 R2016b) which guarantees the validity of results over a continuous range of frequen-364 cies. 365

Finally, note that (2.16) is an NP-hard problem with either pure real or mixed real-complex uncertainties [5], thus all μ algorithms work by searching for upper and lower bounds. The upper bound μ_{UB} provides the maximum size perturbation $\bar{\sigma}(\Delta_u^{UB}) = 1/\mu_{UB}$ for which RS is guaranteed, whereas the lower bound μ_{LB} defines a minimum size perturbation $\bar{\sigma}(\Delta_u^{LB}) = 1/\mu_{LB}$ for which RS is guaranteed to be violated. Along with this information, the lower bound also provides the matrix Δ_u^{LB} determining singularity of the LFT.

3. Main results. In this section the main result of the work is presented. The problem addressed by this article is formally defined in section 3.1 and in section 3.2 a solution by means of a nonlinear optimization program is proposed. The stepby-step presentation, from Program 3.1, which calculates the *smallest* perturbations making the Jacobian unstable, to Program 3.4, which computes the closest subcritical and supercritical Hopf bifurcations, aims at clearly presenting the formulation of robust bifurcation margins. Note that only Program 3.2 and Program 3.4 are actually needed to solve the problem (depending on whether the type of Hopf bifurcation is specified or not). In section 3.3 a multi-start strategy is described, within the extended continuation paradigm, to mitigate the issue of local optima. Finally, in section 3.4 a critical comparison with an alternative method from the literature solving a similar problem is discussed.

3.1. Problem statement. The usual starting point in bifurcation analysis is 385 Eq. (2.1), where f is a nominal vector field, meaning that the only dependence 386 is on the state x and bifurcation parameter p. The latter is of size $n_p = 1$ for 387 continuation of equilibrium points since all their bifurcations have codimension 1, and 388 389 thus 1 parameter is sufficient for its analysis (this of course includes the case of Hopf bifurcations, see Theorem 2.1). Consider the case when parametric uncertainties affect 390 the dynamics, e.g. because of lack of confidence on the values of model parameters or 391 simplifying assumptions underlying the model. The presence of uncertainties can be 392393 modelled by introducing the uncertainty vector δ

394 (3.1)
$$\delta = [\delta_1; ...; \delta_i; ... \delta_{n_\delta}], \quad \delta \in \mathbb{R}^{n_\delta}.$$

The vector field depends now on δ , in addition to x and p. To highlight this, we denote the uncertain vector field by \tilde{f} and the associated Jacobian by \tilde{J}

397 (3.2a)
$$\dot{x} = f(x, p, \delta),$$

398 (3.2b)
$$\tilde{f}: \mathbb{R}^{n_x} \times \mathbb{R} \times \mathbb{R}^{n_\delta} \to \mathbb{R}^{n_x}, \quad \tilde{f} \in \mathcal{C}^{\infty},$$

$$\widetilde{J}: \mathbb{R}^{n_x} \times \mathbb{R} \times \mathbb{R}^{n_\delta} \to \mathbb{R}^{n_x \times n_x}.$$

The objective of the work is then to compute the margins of stable equilibria from the 401 closest Hopf bifurcation for nonlinear systems affected by parametric uncertainties. 402403 To better understand this, assume that the nominal system f has a Hopf bifurcation point (x_H, p_H) , while for another value of the bifurcation parameter \bar{p}_0 a stable fixed 404point \bar{x}_0 exists for f. The goal is to determine the smallest (or worst-case) perturba-405 tion $\bar{\delta} \in \delta$ such that \tilde{f} undergoes a Hopf bifurcation at \bar{p}_0 . It is key to observe that the 406Hopf bifurcation is triggered by perturbations in δ , while the bifurcation parameter 407is fixed at \bar{p}_0 . The reason for this is that the aim here is to compute the margin of 408a certain condition from the occurrence of the bifurcation. Thus, p, which generally 409 defines an operating condition (e.g. load power in an electric power system, speed for 410 an aircraft) is kept fixed at the value \bar{p}_0 which identifies the condition for which the 411 margin is computed. This is different from what is done in the direct method [12] 412 413 (the other approach that looked at a similar problem) where there is no distinction between bifurcation and uncertain parameters, both collected in p (which is then mul-414 tidimensional). As a result of this, all the entries of p are allowed to be perturbed in 415 order to trigger the bifurcation, whereas here the distinction between p (of dimension 416 1) and δ (of dimension n_{δ} , depending on how many uncertainties are considered) is 417 418 clear. See section 3.4 for a thorough comparison with the direct method.

419 It is often relevant to distinguish between supercritical and subcritical Hopf bifurca-420 tions, hence two distinct worst-case perturbations will be considered. For the sake 421 of readability, this distinction will be highlighted in the text when relevant but the 422 notation used will be $\bar{\delta}$ in both cases.

In order to quantify the margin to the closest bifurcation, and thus to allow the concept of worst-case uncertainty to be formalized, a metric for the magnitude of the

perturbation must be adopted. The adopted metric should measure in some quanti-425426tative form the perturbation to which the system is subject. This task is arbitrary and a common approach from robust control is followed [48] (see also section 2.2.2). 427 Consider a generic uncertain parameter d, with w_d indicating the uncertainty level 428 with respect to a nominal value d_0 and $\delta_d \in [-1, 1]$ representing the normalized un-429 certainty range. Note that d_0 and w_d are typically fixed by the analyst based on the 430 knowledge of the nominal value and dispersion of the parameter d respectively. A 431 multiplicative uncertain representation of d is thus obtained as 432

433 (3.3)
$$d = (1 + w_d \delta_d) d_0$$

where $\delta_d = 0$ corresponds to the nominal value of d, while $\delta_d = \pm 1$ represents a 434 perturbation at the extreme of the parameter range (e.g., a variation of $\pm 20\%$ from d_0 435if $w_d = 0.2$). Once the normalization (3.3) is applied to all the uncertain parameters in 436 (3.1), a possible scalar metric (or norm) to quantify the magnitude of the perturbation 437 is the largest of the absolute values of the elements in δ . This can be equivalently 438 expressed as $\bar{\sigma}(\operatorname{diag}(\delta))$, i.e., the maximum singular value of the diagonal matrix with 439elements of δ on the diagonal. Such a metric quantifies the deviation of the uncertain 440 parameters from their nominal values along the direction of the parameter space 441 where this is largest. The objective is thus to compute the perturbation vector with 442 the smallest possible norm, providing therefore the distance from the closest Hopf 443 bifurcation. 444

In fact, $k_m = \bar{\sigma}(\operatorname{diag}(\delta))$ can be regarded as a robust margin from bifurcation be-445cause $k_m \leq 1$ means that a candidate (i.e., within the allowed range of the uncertainty 446 set) perturbation exists which determines a Hopf bifurcation. Thus, the equilibrium 447 \bar{x}_0 of the nominal vector field is not robustly stable at \bar{p}_0 . On the contrary, if $k_m > 1$ 448 then there is no perturbation inside the allowed set which is capable of prompting a 449 Hopf bifurcation. This is pictorially represented in Figure 3, where on the x-axis is 450reported the bifurcation parameter and on the y-axis the margin k_m (note that the 451case $\bar{p}_0 < p_H$ where a Hopf bifurcation is encountered by increasing p is assumed here 452without loss of generality). When the line $k_m = 1$ is crossed, the system is operated in 453a region where Hopf bifurcations can occur in the face of the uncertainties accounted 454

for in the system (shaded area).



FIG. 3. Concept of robust bifurcation margins.

3.2. Solution via nonlinear optimization. The fundamental idea to address 456 457the stated objective is to exploit the interpretation of LFTs discussed in Sec. 2.2.1. Consider for a moment only Condition 1 of Theorem 2.1, which prescribes a pair of 458 purely imaginary eigenvalues for the Jacobian. If J is interpreted as the uncertain 459state-matrix of the linear case, an LFT model of the former with respect to the 460 uncertain parameters in δ can be built up (numerically or analytically [29]). The 461 main difference from the linear case is that in general J is also a function of the states 462 of the system x. This reflects the fact that in the nonlinear context uncertainties 463 have a twofold effect on stability. They directly affect the matrix \hat{J} as independent 464variables, but also indirectly by changing the location of the equilibrium (around 465which the vector field is linearized). The latter is a distinctive feature of the nonlinear 466 467 setting, since in the linear case the location of the equilibrium does not have any effect on the spectrum of the state-matrix, and thus on stability. In full generality, the LFT 468 of the Jacobian $\mathcal{F}(M_{\tilde{I}}, \Delta)$ can be written as 469

470 (3.4a)
$$\mathcal{F}(M_{\tilde{I}}, \Delta) = \mathcal{F}(\mathcal{F}(\mathcal{F}(M_{\tilde{I}}, \Delta_{\nu}), \Delta_{x}), \Delta_{u}),$$

471 (3.4b)
$$\Delta = \operatorname{diag}(\Delta_u, \Delta_x, \Delta_\nu), \quad M_{\tilde{J}} = \begin{bmatrix} M_{\tilde{J}_{11}} M_{\tilde{J}_{12}}; M_{\tilde{J}_{21}} M_{\tilde{J}_{22}} \end{bmatrix}$$

472 (3.4c)
$$\Delta_u = \operatorname{diag}(\delta_1 I_{d_1}, \dots, \delta_i I_{d_i}, \dots, \delta_{n_\delta} I_{d_{n_\delta}})$$

473 (3.4d)
$$\Delta_x = \operatorname{diag}(x_1 I_{x_1}, \dots, x_j I_{x_j}, \dots, x_{n_x} I_{x_{n_x}})$$

474 (3.4e)
$$\Delta_{\nu} = \frac{1}{\nu} I_{n_x}, \quad \nu = i\omega,$$

where (3.4a) exploits the property of interconnected LFTs, and Δ_u is a particular 476 instance of the structured uncertainty set defined in (2.10), considering only real pa-477 rameters. Compared to the linear case (2.15), Δ features now an additional structured 478 block Δ_x , which arises when performing the LFT modeling of \hat{J} due to the states ex-479plicitly appearing in the Jacobian, and for which a similar representation to the one 480for Δ_u is employed. Δ_{ν} finally restricts the attention to purely imaginary eigenvalues 481 of J with frequency ω . 482

Condition 1 of Theorem 2.1 can then be expressed as the singularity of the LFT 483 (3.4a). This is the central step of the proposed extension of μ from the linear con-484 text, where J would be the uncertain state-matrix, to the nonlinear one. In fact, μ 485computes by definition the worst-case perturbation matrix which makes the underly-486 ing LFT ill-posed and employs the same metric (2.16) as the one used to define the 487 robust bifurcation margin k_m . It follows indeed from the definitions and properties 488commented earlier that $k_m = \bar{\sigma}(\operatorname{diag}(\delta)) = \bar{\sigma}(\Delta_u)$. Specifically, k_m is the reciprocal 489 of μ and it has been adopted here because of its straightforward meaning of distance 490491 (or margin) to the onset of a bifurcation. Note in this regard that the symbol k_m was used in the early stages of robust control with the name of excess stability margin 492493 [43, 42].

The discussion above paves the way for the nonlinear program presented next, 494 which aims to compute the smallest perturbation for which \hat{J} has a pair of purely 495imaginary eigenvalues. 496

497

(3.5a)
(3.5b)
(3.5c)

$$\begin{array}{c} \min_{X} k_{m} \quad \text{such that} \begin{cases} \bar{f}(x, \bar{p}_{0}, \delta) = 0, \\ \mathcal{F}(M_{\tilde{J}}, \Delta) \text{ is singular} \\ \bar{\sigma}(\Delta_{u}) \leq k_{m}, \\ \end{cases}$$

$$\begin{array}{c} 497 \\ 498 \\ X = [x; \delta; \omega], \\ \end{array}$$

eters δ ; and frequency ω . \hat{X} will indicate the solution vector gathering \hat{x} , δ , and $\hat{\omega}$ 500

respectively. Let us examine the constraints of the program. Eq. (3.5a) guarantees 501that the solution $(\hat{x}, \hat{\delta})$ corresponds to an equilibrium point for the system. Eq. (3.5b) 502ensures that J has a pair of complex eigenvalues $\nu = \pm \hat{\omega}$, and Eq. (3.5c) bounds the 503

size of the perturbation matrix. 504

This is a similar optimization problem to that in (2.16), with two crucial dif-505 ferences: constraint (3.5a), and the addition of Δ_x in the block Δ of $\mathcal{F}(M_{\bar{i}}, \Delta)$ (to 506 which, notably, constraint (3.5c) does not apply). Due to these differences, available 507 algorithms for μ cannot be applied to compute solutions of (3.5), thus alternative ways 508509should be pursued. Let us examine closely (3.5b), which prescribes singularity of an LFT. According to the definition given in (2.9), necessary and sufficient condition 510for the well-posedness of a generic LFT $\mathcal{F}(M, \Delta_u)$ is the existence of the inverse of 511the matrix $(I - M_{11}\Delta_u)$. Note that M_{11} is, as also previously observed, the transfer matrix seen by the perturbation block Δ_u . In the context of the LFT $\mathcal{F}(M_{\tilde{I}}, \Delta)$ 513514introduced in (3.4), this means that the singularity constraint (3.5b) holds if and only 515if $\det(I - M_{\tilde{J} 11}\Delta) = 0$. This, in turn, can be recast as nonlinear constraints in the optimization variables X. 516

As for (3.5c), this is a non-smooth constraint because of the maximum singular value 517 operator, but it can be drastically simplified by exploiting the structure of Δ_u (3.4c). 518Indeed this constraint is equivalent to 519

520 (3.6)
$$-k_m \le \delta_i \le k_m, \quad i = 1, ..., n_{\delta},$$

which is a set of linear inequalities in the optimization variables and the objective 521function k_m . Note that a similar relaxation would hold also for complex scalar uncer-

tainties, not considered in this work. 523

Based on the previous discussion, the following smooth nonlinear optimization 524 problem is proposed to solve Program 3.1. 525

Program 3.2.

(3.7a)
(3.7b)
$$\min_{X} k_{m} \text{ such that} \begin{cases} \tilde{f}(x, \bar{p}_{0}, \delta) = 0, \\ \det(I - M_{\tilde{f} \ 11} \Delta) = 0, \\ -k_{m} < \delta_{i} < k_{m}, \quad i = 1, ..., n_{\delta}, \end{cases}$$

526 527

 $X = [x; \delta; \omega], \quad n_{ctrs} = n_x + 2 + n_\delta,$

where n_{ctrs} denotes the number of total constraints of the optimization. 528

The key idea behind Program 3.2 is to enforce singularity of the LFT (3.5b) by using directly the determinant condition represented by constraint (3.7b). In [40] 530531 this is listed among the known methods for the computation of μ_{LB} , and examples of related algorithms can be found in [22, 47]. The approaches presented in those 532works, however, are limited to the case of linear systems, i.e., they represent alter-534natives to well-established μ lower bounds algorithms such as the power iteration [36] and the gain-based method [44]. To the best of the authors' knowledge, this 535 536 is indeed the first time that the concept of structured singular value is used in the context of worst-case bifurcations of a nonlinear vector field. Moreover, Program 3.2 recasts the optimization so that the objective function and the constraints are smooth. 538 This differs from the aforementioned works where the optimization was performed by 539minimizing the nonsmooth function $\bar{\sigma}(\Delta_u)$. This is overcome here by considering 540

the relaxation commented in (3.6) and introducing the objective function k_m as an additional optimization variable.

543 Remark 3.1. Constraint (3.7b) consists of two (real and imaginary parts of the 544 determinant) nonlinear equality constraints in the variables X. By using Laplace 545 expansion of the determinant [1] and the fact that Δ is structured, an analytical 546 expression for the gradient of (3.7b) with respect to δ and x can be obtained and 547 provided to the optimizer. As for ω , this is more tedious and therefore finite differences 548 are employed.

Note also that, from a continuation perspective, (3.7b) can be regarded as an analog 549of the real scalar test functions commonly used to detect Hopf bifurcations [3]. The 550latter can be efficiently formulated by means of bordered matrices techniques and 551have the property that the test function has a zero at a bifurcation point. The main difference here is that (3.7b) is complex, thus consists of two real scalar equations. 553 This is due to the fact that the frequency ω of the purely imaginary eigenvalues appear 554explicitly in the constraint (and thus is an additional independent variable), which is different from the test functions formulation. This is an important feature of the 556557developed approach, and possible ways to exploit it will be discussed later.

558 Enforcing the transversality condition

Program 3.2 allows worst-case perturbations to be computed such that the Ja-559 cobian of f linearized around the perturbed equilibrium point has a pair of purely 560imaginary eigenvalues. This, however, does not guarantee that the perturbed system 561 undergoes a Hopf bifurcation because transversality (Condition 2 of Theorem 2.1) is 562 not automatically verified. Constraints guaranteeing that transversality is satisfied 563 can be appended to (3.7) in different ways, including using test functions [3] or au-564tomatic differentiation [23]. Here an approach leveraging the versatility of the LFT 565 paradigm is proposed. Consider a small fixed constant ϵ_p which defines the perturbed 566 bifurcation parameter $\bar{p}_{\epsilon_p} = (1 + \epsilon_p)\bar{p}_0$. The LFT $\mathcal{F}(M^{\epsilon}_{\tilde{j}}, \Delta^{\epsilon})$ of the Jacobian at \bar{p}_{ϵ_p} 567 can be written following (3.4) as 568

569 (3.8a)
$$\mathcal{F}(M^{\epsilon}_{\tilde{J}}, \Delta^{\epsilon}) = \mathcal{F}(\mathcal{F}(\mathcal{F}(M^{\epsilon}_{\tilde{J}}, \Delta^{\epsilon}_{\nu}), \Delta^{\epsilon}_{x}), \Delta_{u}),$$

570 (3.8b) $\Delta^{\epsilon} = \operatorname{diag}(\Delta_{u}, \Delta_{x}^{\epsilon}, \Delta_{\nu}^{\epsilon}), \quad M_{\tilde{J}}^{\epsilon} = \left[M_{\tilde{J}_{11}}^{\epsilon} M_{\tilde{J}_{12}}^{\epsilon}, M_{\tilde{J}_{21}}^{\epsilon} M_{\tilde{J}_{22}}^{\epsilon}\right],$

571 (3.8c)
$$\Delta_x^{\epsilon} = \operatorname{diag}((1+\epsilon_x)x_1I_{x_1}, ..., (1+\epsilon_x)x_jI_{x_j}, ..., (1+\epsilon_x)x_{n_x}I_{k_{n_x}}),$$

572 (3.8d)
$$\Delta_{\nu}^{\epsilon} = \frac{1}{\nu^{\epsilon}} I_{n_x}, \quad \nu^{\epsilon} = \epsilon_{\nu} + (1 + \epsilon_{\omega})\omega,$$

574 where ϵ_{ν} , ϵ_x , and ϵ_{ω} are unknown scalars described later. The following optimization 575 problem is then proposed to determine the worst-case perturbation for which both 576 conditions of the Hopf theorem are guaranteed to hold, that is, to calculate the margins 577 to the closest Hopf bifurcation point.

Program 3.3.

(3.9a)
$$\int \tilde{f}(x,\bar{p}_0,\delta) = 0,$$

(3.9b)
$$\det(I - M_{\tilde{i} 11}\Delta) = 0$$

(3.9c)
$$\min k_m$$
 such that $\begin{cases} -k_m \le \delta_i \le k_m, \end{cases}$

(3.9d)
$$\tilde{f}((1+\epsilon_x)x,\bar{p}_{\epsilon_p},\delta) = 0$$

(3.9e)
$$\det(I - M^{\epsilon}_{\tilde{I} \ 11} \Delta^{\epsilon}) = 0,$$

$$X = [x; \delta; \omega; \epsilon_{\nu}; \epsilon_x; \epsilon_{\omega}], \quad n_{ctrs} = n_x + 2 + n_\delta + n_x + 2$$

Underlying Program 3.3 there is a perturbation argument which builds on the application of the IFT to the states x and the eigenvalue ν of the vector field $\tilde{f}(x, p, \hat{\delta})$ for fixed $\hat{\delta}$ and p in a neighbourhood of \bar{p}_0 . Indeed, at $p = \bar{p}_0$, it holds $x = \hat{x}$ and $\nu = i\hat{\omega}$ for the constraints (3.9a-3.9c). When perturbing p by a small increment ϵ_p , a first order approximation for x and ν is assumed, and reflected in the choice of the scalars ϵ_x (3.8c), as well as ϵ_{ω} and ϵ_{ν} (3.8d). A vector ϵ_x , with an element for each component of x, could also be considered, by adding $n_x - 1$ unknowns to Program 3.3.

Remark 3.2. Program 3.2 does not mathematically guarantee the onset of a Hopf bifurcation because it does not take into account the transversality condition, and 593for this reason Program 3.3 is proposed. However, for engineering systems where p594has a physical meaning (e.g., load power in a power system, speed for an aircraft) 596the transversality condition is often automatically verified. In fact, cases where this condition is not satisfied are termed *degenerate* in the literature [18]. For this reason, 597 the problem was stated in Sec. 3.1 assuming that the nominal system has a bifurcation 598 at p_H whereas for $p = \bar{p}_0$ the system has a stable equilibrium. It is thus implicit in 599the formulation of the problem that a change of p has an effect on the stability of 600 the system. In particular, it is expected that the critical eigenvalues of the perturbed 601 602 Jacobian will cross the imaginary axis as p is perturbed around \bar{p}_0 .

It is observed that, compared to Program 3.2, Program 3.3 only adds three unknowns to the vector of optimization variables X, and has $n_x + 2$ additional constraints. Its effect in terms of computational cost is thus not expected to be important.

However, a strong reason to resort to Program 3.2 whenever possible is related to 606 607 the local optimality of the solutions of nonlinear programs. This issue will be further discussed in Sec. 3.3, but it is remarked here that the addition of the constraints 608 (3.9d-3.9e) has a detrimental effect on it. Indeed it is always advisable in nonlinear 609 optimization to avoid redundant constraints in order to reduce the likelihood of local 610 611 optima [33]. Based on these considerations, and the discussion in Remark 3.2, the proposed strategy is to employ Program 3.2 to find robust bifurcation margins and, if 612 continuation analyses of the perturbed system show that the transversality condition 613 is not fulfilled, use Program 3.3. It is noted that none of the analyses done in support 614 of this study required the adoption of Program 3.3 (which however was tested to verify 615 its soundness). For this reason, and also for the sake of clarity, in the remainder of the 616 617 work Program 3.2 will be considered as the basis for discussion and further algorithms.

618 Specifying the type of closest Hopf bifurcation

The robust bifurcation margin k_m has been associated so far with the occur-619 rence of a generic Hopf bifurcation. Attention is now focused on the nature of the 620 bifurcation, i.e., subcritical or supercritical. The idea is to add a condition on the 621 622 sign of the Lyapunov coefficient l_1 to the constraints of Program 3.2. This can be done by using the definition of l_1 (2.4), which requires the computation of left and 623 right eigenvectors associated with the critical eigenvalue, and the tensors of second 624 and third order derivative. By exploiting the fact that ω is an optimization variable, 625626 the eigenvectors can be computed without performing an eigenvalue analysis, but by

direct computation as follows 627

(3.10)

$$(J - i\omega I_{n_x})q = 0, \quad q = [1; q_l],$$

$$(\tilde{J}^T + i\omega I_{n_x})r = 0, \quad r = [1; r_l],$$

$$\langle r, q \rangle = 1,$$

where without loss of generality the first element of the eigenvectors has been fixed 629 to 1. As for the tensors, the derivatives in (2.6) can be computed analytically in 630 simple cases and by automatic or symbolic differentiation for more complex ones. 631 Alternatively, in [27] efficient strategies to avoid computing second and third order 632 633 derivatives of the vector field are discussed. In any case, they are available as a function of the optimization variables x and δ , and thus the only addition to the 634 vector of unknowns X is essentially l_1 . 635

In conclusion, given a positive tolerance ϵ_l on the value of the Lyapunov coefficient, 636 and an integer $s_l = \pm 1$ defining the sign of l_1 ($s_l = 1$ for subcritical and $s_l = -1$ 637

Program 3.4.

(3.11a)
(3.11b)
$$\min k_{---} \text{ such that } \begin{cases} \tilde{f}(x, \bar{p}_0, \delta) = 0, \\ \det(I - M_{\tilde{J} \ 11} \Delta) = 0, \end{cases}$$

(3.11c)
$$\begin{array}{c} \min_{X} k_m \quad \text{such that} \\ (3.11d) \\ \end{array} \begin{cases} -k_m \leq \delta_i \leq k_m, \quad i = 1, ..., n_{\delta}, \\ s_l l_1 - \epsilon_l > 0, \\ \end{array} \end{cases}$$

$$X = [x; \delta; \omega; l_1], \quad n_{ctrs} = n_x + 2 + n_\delta + 1.$$

To summarize the content of this Section, the problem of computing the closest 642 643 Hopf bifurcation point in the uncertain parameter space has been formulated via a nonlinear optimization problem and has been presented incrementally in order to 644 645 stress the key steps involved. Because the Hopf bifurcation can be of two types, namely subcritical and supercritical, two Programs are proposed. Program 3.2 determines 646 the closest Hopf bifurcation to a given stable equilibrium (this might be subcritical or 647 supercritical, depending on the specific case), whereas Program 3.4 allows the type of 648 649 closest Hopf bifurcation (via a constraint on the Lyapunov coefficient) to be specified. 650

3.3. Continuation-based multi-start strategy. The programs discussed in 651 Section 3.2 allow margins to Hopf bifurcation for a nominally stable equilibrium point 652 653 in the face of uncertainties to be computed. The main issue with this approach is that, due to the fact that is based on nonlinear optimization, there is no guarantee 654 655 that the one found is the closest bifurcation, and thus in practice only upper bounds on k_m are computed. In other words, global minima might be missed and thus there 656 could be a vector $\overline{\delta}$ featuring a smaller norm than $\widehat{\delta}$ which causes a Hopf bifurcation. 657 658 Local optima are a well known issue in nonlinear optimization and, while there exist global optimization algorithms that can guarantee global optima, their computational 659 660 burden grows exponentially with the dimension of the problem and thus often are not practical solutions [33]. 661

Mitigation strategies when local solvers (e.g. interior point methods) are used 662 depend on several aspects, including specific features of the program (e.g., objective 663664 functions) and adopted optimization algorithms [17]. For this problem the objective

16

is to compute worst-case perturbations quantified by means of a scalar metric, thus a possible way to account for this issue is to estimate a guaranteed smallest magnitude of the perturbation for which the system is stable. This is the approach taken in μ analysis, where the computation of μ_{LB} is known to be prone to local minima and as a remedy upper bounds μ_{UB} have been proposed. Lower bounds on k_m (nonlinear analogs of μ_{UB}) could then be a strategy in the present context, but this has not been pursued here and could be a topic of future research.

As for the optimization algorithms, the focus of this work is not on developing 672 ad-hoc advanced optimization strategies, hence off-the-shelf algorithms available in 673 MATLAB for nonlinear constrained problems are employed [31]: These include: *in*-674 terior point, which solves the constrained problem using a sequence of unconstrained 675 676 optimizations by using barrier or penalty functions to account for the constraints; active set and sqp, belonging to the class of sequential quadratic programmes, which 677 directly solve the constrained problem via a series of approximating quadratic pro-678 gramming based on the Karush-Kuhn-Tucker equations (necessary conditions for op-679 timality of constrained optimization problems). Leveraging the availability of solvers 680 681 based on different optimization methods, a (naive but possible) strategy employed in the work is to restart the programs using different solvers. 682

Another good practice to reduce the likelihood of local minima is to formulate the 683 problem in the *simplest* way possible [33], e.g., using smooth objective functions and 684 constraints and avoiding redundant constraints. These two principles have guided 685 the idea of introducing the objective function k_m to relax the non-smooth bound 686 687 on the uncertainty set involving $\bar{\sigma}$, which lead to the equivalent constraints (3.6). Moreover, the aim of simplifying as much as possible the set of constraints prompted 688 the discussion in Remark 3.2, where it was proposed (based on a physically moti-689 vated assumption) to resort to Program 3.3 only if the solution does not satisfy the 690 transversality condition. 691

A strategy which exploits a distinctive feature of this formulation is to run Pro-692 693 gram 3.2 at a given frequency, i.e., ω does not belong to X but is fixed a priori. The rationale behind this is twofold. From a mathematical point of view, the optimization 694 is simplified by the fact that constraint (3.7b) does not depend on the frequency and 695 this enhances the accuracy of the result. From a bifurcation perspective, fixing the 696 frequency restricts the mechanisms by which the system can undergo a Hopf bifur-697 cation when subject to uncertainties, which reduces the number of feasible solutions 698 in the first place, and as a result makes it also more likely to detect the optimal one. 699 A value of k_m can be associated with each discrete frequency, and the smallest of 700 these values can be regarded as the most critical. A natural drawback of this ap-701 proach is that critical frequencies can be missed, but this can be overcome by running 702 703 Program 3.2 in a second step with ω as optimization variable and initializing it with 704 values corresponding to the critical solution.

Despite these measures, the risk of falling into local minima is still present. In particular, the programs' initialization represents a critical aspect and thus a continuation-based multi-start strategy is proposed. Assume that the optimizer has found a solution \hat{X} to Program 3.2. The goal is then to provide the optimizer with a set of initializations, derived from \hat{X} but possibly not leading the optimizer to find the same solution, which allows an exhaustive optimization campaign to be performed. 712 is first considered

713 (3.12)
$$F(x,\delta,\omega,\lambda_d,\lambda_k) = \begin{pmatrix} \tilde{f}(x,\bar{p}_0,\delta) \\ \det(I-M_{\tilde{J}\ 11}\Delta) \\ \bar{\sigma}(\Delta_u) \end{pmatrix} - \begin{pmatrix} 0 \\ \lambda_d \\ \lambda_k \end{pmatrix} = 0.$$

This can be recast in the formalism of (2.7) by setting

$$u = X = [x; \delta; \omega], \ u \in \mathbb{R}^{n_u}, \ n_u = n_x + n_\delta + 1,$$

$$\lambda = [\lambda_d; \lambda_k], \ \lambda_d \in \mathbb{R}^2, \ \lambda_k \in \mathbb{R}^1,$$

715 (3.13)

$$\Phi = \tilde{f}(x, \bar{p}_0, \delta), \ \Phi : \mathbb{R}^{n_u} \to \mathbb{R}^{n_x},$$

$$\Psi = [\det(I - M_{\tilde{J}\ 11}\Delta), \bar{\sigma}(\Delta_u)], \ \Psi : \mathbb{R}^{n_u} \to \mathbb{R}^3$$

$$F : \mathbb{R}^{n_x + n_\delta + 3} \to \mathbb{R}^{n_x + 3}.$$

Let $\mathbb{I} = \{1, 2\}$ and $\overline{\mathbb{I}} = \{3\}$ be its complement, with $\lambda_{\mathbb{I}} = \{\lambda_i | i \in \mathbb{I}\}$ and $\lambda_{\overline{\mathbb{I}}} = \{\lambda_i | i \in \mathbb{I}\}$ 716 $\bar{\mathbb{I}}$, and $u^* = \hat{X}$, $\lambda^* = \Psi(u^*)$. By construction, the restriction $F(u^*,\lambda)|_{\lambda_{\mathbb{I}}=\lambda^*_*} = 0$ 717and $F(u,\lambda)|_{\lambda_{\mathbb{I}}=\lambda_{\mathbb{I}}^*}$ satisfies the IFT at (u^*,λ^*) . Therefore, $F(u,\lambda)|_{\lambda_{\mathbb{I}}=\lambda_{\mathbb{I}}^*}$ defines a 718continuation problem for the *d*-manifold with $d = n_x + n_{\delta} + 1 - (n_x + 2) = n_{\delta} - 1$. 719 720 Note that $\lambda_{\mathbb{I}}$ (coinciding with λ_d) are inactive continuation parameters (corresponding 721 to active constraints) because they are kept constant during continuation and they ensure the singularity of the LFT $\mathcal{F}(M_{\tilde{J}}, \Delta)$. Since $\lambda_d^* = 0$, the corresponding active 722 constraints could have been equivalently embedded in the zero function Φ but, for 723consistency with the parallel between f and Φ discussed in Sec. 2.1.2, this has been 724 used for the vector field only. On the other hand, λ_{π}^{*} (i.e., λ_{k}) corresponds to an 725 726 inactive monitor function bookkeeping the magnitude of the perturbation at each step of the continuation. 727

The manifold associated with (3.12), denoted here by \mathcal{H} , represents the set of Hopf bifurcation points *connected* to the solution \hat{X} in the uncertain parameter space. A first important observation is that the dimension of \mathcal{H} is $n_{\delta} - 1$. This is in agreement with the well known fact [3] that a branch (i.e., 1-dimensional manifold) of Hopf points can be obtained by continuing simultaneously two parameters starting from a known initial point. Indeed, in the case of two uncertainties $(n_{\delta} = 2) \mathcal{H}$ is the branch of Hopf points connected to the initial solution \hat{X} .

In principle, the computation of \mathcal{H} could directly locate bifurcation points associ-735 ated with perturbations featuring a smaller magnitude than $\hat{\delta}$ by monitoring λ_k (note 736 however that they could still be local optima since only the connected branches can 737 be tracked). In addition to that, exploring the surroundings of X (using a continu-738 ation meaning of this terminology) can provide the sought initialization points for a 739 new optimization campaign. Unfortunately, \mathcal{H} is generally multidimensional. In fact, 740 741 it is reasonable to assume that even for a relatively small number of uncertainties 742 computing \mathcal{H} is not viable. To overcome this, a 1-dimensional restriction of \mathcal{H} is constructed by considering a parametrization of the uncertainty set δ with a vector 743 function $g(z,y): \mathbb{R}^2 \to \mathbb{R}^{n_\delta}$, where the 2 independent variables z and y have been 744 introduced. The definition of g is arbitrary and various strategies can be pursued. 745The approach taken here assumes that two solutions \hat{X}^1 , and \hat{X}^2 from Program 3.2 746are available (their selection will be commented on later). Given the associated per-747

18

turbation vectors $\hat{\delta}^1$, and $\hat{\delta}^2 \in \mathbb{R}^{n_\delta}$, a possible choice for g is then

749 (3.14)
$$g(z,y): \mathbb{R}^2 \to \mathbb{R}^{n_\delta} \begin{cases} \hat{\delta}_1^1 z + \hat{\delta}_1^2 (1-y), \\ ... \\ \hat{\delta}_i^1 z + \hat{\delta}_i^2 (1-y), \\ ... \\ \hat{\delta}_{n_\delta}^1 z + \hat{\delta}_{n_\delta}^2 (1-y), \end{cases}$$

Note that by construction $g(1,1) = \hat{\delta}^1$ and $g(0,0) = \hat{\delta}^2$.

751 Based on this, the following continuation problem is formulated

752 (3.15)
$$F(x,\delta,\omega,z,y,\lambda_d,\lambda_k,\lambda_g) = \begin{pmatrix} \tilde{f}(x,\bar{p}_0,\delta) \\ \det(I-M_{\tilde{f}\ 11}\Delta) \\ \bar{\sigma}(\Delta_u) \\ \delta - g(z,y) \end{pmatrix} - \begin{pmatrix} 0 \\ \lambda_d \\ \lambda_k \\ \lambda_g \end{pmatrix} = 0.$$

With respect to the definitions in (3.13), z and y have been added to the vector of continuation variables u (i.e., u = [X; z; y]), while the vector function $\delta - g$ has been added to the family of monitor functions Ψ (with associated continuation parameters $\lambda_g \in \mathbb{R}^{n_\delta}$).

The Let $\mathbb{I} = \{1, 2, 4, ..., 4 + n_{\delta}\}$, and $\overline{\mathbb{I}}$, $\lambda_{\mathbb{I}}$, $\lambda_{\overline{\mathbb{I}}}$ as before. Two starting points are available, respectively $u^* = [\hat{X}^1; 1; 1]$ and $u^* = [\hat{X}^2; 0; 0]$, with $\lambda^* = \Psi(u^*)$. Note that in both cases $\lambda_g^* = \Psi(u^*) = 0$ by construction. Therefore, $\delta = g(z, y)$ at each step of the continuation, and δ is expressed as a linear combination of $\hat{\delta}^1$ and $\hat{\delta}^2$.

Since $F(u^*, \lambda)|_{\lambda_{\mathbb{I}}=\lambda_{\mathbb{I}}^*} = 0$ and $F(u, \lambda)|_{\lambda_{\mathbb{I}}=\lambda_{\mathbb{I}}^*}$ satisfies the IFT at (u^*, λ^*) , then a manifold \mathcal{H}_g with dimension $d = n_x + n_\delta + 3 - (n_x + 2 + n_\delta) = 1$ is defined. Crucially, the dimension is 1 irrespective of the number of uncertainties n_δ , with the drawback that these are now constrained to vary according to (3.14). \mathcal{H}_g^1 and \mathcal{H}_g^2 indicate the manifold built starting from $[\hat{X}^1; 1; 1]$ and $[\hat{X}^2; 0; 0]$ respectively, with the subscript and the superscript highlighting the dependence on the parametrization of the uncertainties g and the initial point.

The construction of \mathcal{H}_g requires two perturbation vectors $\hat{\delta}^1$ and $\hat{\delta}^2$. This is not 768 restrictive, since as a result of the local optimality typically more than one solution 769is available. In addition, the possibility of running the optimization at a fixed fre-770 quency ω can be advantageously exploited with the goal of obtaining different modes 771 772 of perturbations. Indeed, as discussed before, Hopf bifurcations occurring at different frequencies could represent different mechanisms underlying the loss of stability, thus 773 considering a linear combination of the perturbations as in (3.14) represent an efficient 774 strategy to select points on \mathcal{H}_q . 775

To sum up the multi-start strategy approach, the starting point is Program 3.2 776 which provides a solution consisting of an equilibrium point \hat{x} of f perturbed by δ such 777 that the associated Jacobian J has a pair of purely imaginary eigenvalues. This is not 778 necessarily the closest bifurcation point to the nominal system due to the possibility 779 of local minima. However, \hat{X} can be used to compute the restricted manifold \mathcal{H}_g via a 780 numerically cheap continuation problem once a parametrization g for the uncertainty 781 set is provided. Continuation of \mathcal{H}_q has two objectives. First, it could directly detect 782 improved solutions of Program 3.2 (if $\lambda_k < \hat{k}_m$). Second, points on \mathcal{H}_q can be used 783 to run Program 3.2 with different initializations. 784

If the manifold \mathcal{H}_g gathers a large number of points, and running the optimization for each of them is not viable, criteria could be employed to select a subset of them only. Keeping in mind that the goal is to provide initializations which possibly make the optimizer converge to different points from the initial solution \hat{X} , the premise of these criteria is to detect on \mathcal{H}_g perturbation vectors *qualitatively* different from $\hat{\delta}$. Possible indicators are for example the frequency ω and the changes in sign of the parameters in δ (recall that these are normalized, thus a change in sign reveals a change in the direction of perturbation for the considered parameter).

3.4. Comparison with the direct method. The framework presented in the 793 794 previous sections allows the computation of the robust bifurcation margin k_m via nonlinear optimization (section 3.2) aided by a multi-start strategy (3.3). Despite its 795importance for the analysis of nonlinear systems, the computation of the closest Hopf 796 bifurcation point to a stable equilibrium in the uncertain parameter space has not 797 798 been adequately investigated so far. The only alternative approach available in the literature is the so-called *direct* method [12], and the objective of this section is to 799 point out the differences (and the associated advantages) of the formulation proposed 800 in this paper (in the remainder of this section termed *margin* method) with respect 801 802 to it.

The direct method for Hopf bifurcations considers as starting point the vector 803 804 field (2.1) where $n_p > 1$, i.e. the vector of bifurcation parameters is multidimensional. Given a vector \bar{p}_0 associated with a stable equilibrium, the closest point to \bar{p}_0 in the 805 set of parameters (or hypersurface) Σ for which the equilibrium experiences a Hopf 806 bifurcation is sought. A first difference is thus that in the margin method a distinction 807 is drawn between bifurcation parameter p (of dimension equal to the codimension of 808 809 the bifurcation, which is 1 for the Hopf case) and uncertain parameters δ , and the 810 closest Hopf point is sought in the uncertainty space only (that is, \bar{p}_0 is fixed). Conversely, in the direct method bifurcation and uncertain parameters are all gathered 811 in p and can all be perturbed in order to reach the closest bifurcation point. This 812 difference only pertains to the formulation of the problem, but it is worth highlighting 813 it since two different perturbation scenarios are effectively considered. 814

The key observation leveraged by the direct method is that if p_* is the closest point to \bar{p}_0 in Σ , then the vector $p_* - \bar{p}_0$ is parallel to the normal vector to the hypersurface Σ at p_* . Moreover, p_* is a *local* minimum if the distance $|p_* - \bar{p}_0|$ is smaller than the reciprocal of the curvature of Σ at p_* .

Implementation of these conditions lead to the extended system of equations defining 819 a Hopf bifurcation ([12], Section 5). The name *extended* derives from the fact that, for 820 821 $n_p = 1$, this set of equations reduces to the standard system of equations to compute Hopf bifurcation branches (Th. 2.1). The multidimensional case exploits the fact that 822 the normal vector at p_* can be written out as a function of $\nabla_p f|_{p=p_*}$ and of the eigen-823 vector of the Jacobian $\nabla_x f|_{p=p_*}$ associated with the purely imaginary eigenvalues. 824 In turn, the curvature can be written as a function of the normal vector. Building 825 826 on these relationships and enforcing all the associated constraints, the problem is finally formulated as the solution of $6n_x + n_p + 2$ nonlinear equations in $6n_x + n_p + 2$ 827 unknowns. Similarly to the margin method (see the vector X in Program 3.2), the 828 unknowns of the problem include the perturbed equilibrium (n_x) , the closest bifur-829 cation parameter vector (n_p) , and the frequency (1). However, in addition to these 830 831 there are another $5n_x + 1$ unknowns which are introduced in order to express the rest of the constraints, and clearly do not feature in the margin method. The key ideas 832 leveraged by the margin method to avoid these additional constraints are to enforce 833 the constraint on the Jacobian as singularity of the LFT (3.7b) and cast the minimum 834 distance problem as maximum singular value minimization of the perturbation matrix 835

- 836 Δ . As for the number of constraints, Program 3.2 has $n_{ctrs} = n_x + 2 + n_\delta$ while the 837 direct method features $6n_x + n_p + 2$. A comparison in terms of size of the problem, 838 both in terms of unknowns and constraints, points out an objective advantage of the 839 margin method with respect to the direct method. Quoting the author in [12], "this 840 direct method for computing Hopf bifurcations may be too cumbersome to be useful if 841 n_x is large".
- The distinctions between the two methods are however not restricted to the size of the 842 problem. For example, the mathematical formulation of the problem is different. In 843 the margin method, k_m is the result of an optimization problem whereas in the direct 844 method a determined (the number of constraints equals the number of unknowns) set 845 of nonlinear equations has to be solved (e.g. with Newton-type methods). This is 846 847 deemed an advantage of the margin method, since the greater degree of freedom in finding the solution can be exploited using optimization techniques in order to achieve 848 higher efficiency in the computation and more robustness to the problem of local min-849 ima. As for the latter aspect, it is further observed that the margin method is also 850 equipped with the multi-start strategy (3.3), as opposed to the direct method where 851 there are no strategies to directly tackle the problem of converging to local minima. 852 853 Another favourable feature offered by the margin method is that it allows the type of closest bifurcation to be specified via constraint on the Lyapunov coefficient (Pro-854 gram 3.4). This is done in a relatively straightforward way by using the fact that ω 855 is an optimization variable, and thus the eigenvectors needed for the computation of 856 l_1 (2.4) are available without performing an eigenvalue analysis (3.10). As a result, 857 858 Program 3.4 only adds one unknown (l_1) and one scalar constraint to Program 3.2 where the type of bifurcation is not specified. This is again due to the LFT formula-859 tion of the problem that provides an analytic dependence of the constraints on ω (see 860 also Remark 3.1). Conversely, the option of specifying the closest bifurcation is not 861 available in the direct method, nor is it clear how it could be added without incurring 862 a further substantial increase in the number of unknowns and constraints. 863

Another important aspect is related to the type of constraints involved in the two problems. As discussed in Remark 3.1, the gradients of the constraints in Program 3.2 with respect to the unknowns (with the exception of ω , which is more tedious) can all be analytically computed and provided to the solver, with great advantage in terms of efficiency of computation. This clearly does not apply to the direct method due to the very complicated definition of the constraints (involving eigenvectors and their projections) and of the unknowns.

Finally, a unique feature of the robust bifurcation margin k_m owes to its interpre-871 tation as nonlinear extension of the structured singular value μ . This indeed opens 872 up the possibility to transfer to the bifurcation field many of the well established 873 874 approaches in robust control [48]. This applies to: modelling, where advanced LFT algorithms [28, 29] can be employed to efficiently formulate the constraints of Pro-875 gram 3.2 and Program 3.4; analysis, where the insightful interpretations of μ and 876 associated analysis strategies (sensitivity, frequency-domain analysis) [26] carry over 877 to k_m ; and ultimately robust control design, whereby a (potentially nonlinear) con-878 879 troller is synthesised to prevent bifurcations in the face of a given uncertainty set. While examples of the first two aspects have been given throughout the section and 880 881 will be exemplified further in section 4, the latter is an exciting prospective line of research that can build on this initial work. 882

4. Numerical examples. The proposed concept of robust bifurcation margin is demonstrated on two test cases from the literature. The first is a power system model for which the sensitivity of the Hopf bifurcation to modeling parameters was considered in [13], while the second is an aeroelastic case study previously studied with linear robust control techniques in [25].

4.1. Power system.

4.1.1. Model description. The first example considers the single machine 889 power load system with voltage regulator and dynamic load model studied in [13] 890 and depicted in Fig. 4. The model used in [13] is very similar to the one originally 891 proposed in [7], with the variations discussed next. The model in [7] consists of: five 892 ordinary differential equations representing the dynamics of the generator voltages 893 $E'_{d} + jE'_{q}$, the voltage regulator state R_{f} and output voltage V_{R} , and the field voltage 894 E_{FD} ; two algebraic equations which relate the load bus voltage phasor V_L/θ to the 895 voltage source $E'_d + jE'_s$ and the load demand $P_L + jQ_L$, where P_L and Q_L are respec-896 tively the constant (and fixed a priori) active and reactive power components. The 897goal of the regulator is to control the voltage E_s at the high side of the transformer 898 given a reference voltage setpoint E_{ref} , which depends on the loading level.



FIG. 4. Power system sketch.

899

Differently from [7], the model in reference [13] considers: a dynamic power load (i.e. P_L and Q_L are not constant); a setpoint E_{ref} which is fixed for all loading levels; and an expression of the voltage E_s as a function of the other state variables. Due to these changes, two ordinary differential equations are added for V_L and θ , and the two algebraic equations become explicit equations for P_L and Q_L .

⁹⁰⁵ The resulting set of seven ordinary differential equations describing the power system, ⁹⁰⁶ with vector of states $x = [E'_d; E'_a; V_R; E_{FD}; R_f; \theta; V_L]$, is:

907 (4.1a)
$$T'_{q0}\dot{E}'_{d} = -E'_{d} + (x_q - x'_d)I_q,$$

908 (4.1b)
$$T'_{d0}\dot{E}'_{q} = -E'_{q} - (x_d - x'_d)I_d + E_{FD},$$

909 (4.1c)
$$T_A \dot{V}_R = -V_R + K_A (E_{ref} - E_s - \frac{K_f E_{FD}}{T_f} + R_f),$$

910 (4.1d)
$$T_E E_{FD} = -E_{FD} + V_R,$$

911 (4.1e)
$$T_f \dot{R}_f = -R_f + \frac{K_f E_{FD}}{T_f},$$

912 (4.1f)
$$D\dot{\theta} = P_L - lPF,$$

913 (4.1g)
$$k\dot{V}_L = Q_L - l\sqrt{1 - PF^2}.$$

where: T_{q0} and T_{d0} are the open circuit time constants; x_d and x_q are the synchronous reactances; x'_d is the transient reactance; I_d and I_q are the currents; T_A and K_A are the voltage regulator time constant and gain; T_E is the exciter time constant; T_f and K_f are the time constant and gain of the feedback loop; D and k are time constants of the load dynamics; PF is the power factor and l parameterizes the increase of the 920 constant power part of the load (this will be used as bifurcation parameter in the 921 analyses).

This set of equations must be closed with the defining equations for I_d , I_q , P_L , Q_L , and E_s . For the currents, the following holds [7]:

924 (4.2)
$$I_{d} = \frac{1}{x_{E}} (E'_{q} - V_{L} \cos(\delta - \theta)),$$
$$I_{q} = \frac{1}{x_{E}} (-E'_{d} + V_{L} \sin(\delta - \theta)),$$
$$x_{E} = x'_{d} + x_{T} + x_{e}.$$

where δ is the rotor angle, x_T is the high side transformer reactance and x_e is the transmission line reactance.

⁹²⁷ The equations for the remaining three variables are not provided in [13]. The relationships for P_L and Q_L are derived here from the two aforementioned algebraic equations in [7], which now allow an explicit expression for the load components to be obtained since the phasor $V_L/\underline{\theta}$ has a dedicated dynamic description (4.1f-4.1g). As for E_s , a relationship to the state variables is derived by considering the loadflow equation for the circuit with the voltage source at the high side of the transformer. This leads to:

934 (4.3a)
$$P_L = \frac{V_L}{x_E} \cos(\theta) \tilde{P} - \frac{V_L}{x_E} \sin(\theta) \tilde{Q},$$

935 (4.3b)
$$Q_L = \frac{V_L}{x_E} \sin(\theta) \tilde{P} + \frac{V_L}{x_E} \cos(\theta) \tilde{Q},$$

936
$$\tilde{P} = -E'_d \cos(\delta) + E'_q \sin(\delta) - V_L \sin(\theta)$$

937
$$\tilde{Q} = E'_{d}\sin(\delta) + E'_{q}\cos(\delta) - V_{L}\cos(\theta),$$

938 (4.3c)
$$E_s = \frac{1}{V_L} \sqrt{(x_e P_L)^2 + (x_e Q_L + V_L^2)^2}.$$

Note that the same expression for E_s was used in [46], where a very similar power system was analyzed.

Table 1 reports the values of the parameters used here for the power system model. These are all taken from [7], except for D and k (introduced anew in [13]) and K_f , whose value was changed in [13]. As for the rotor angle δ , it is noted that their dynamic is assumed faster than the dominant voltage dynamics, thus the angle is in quasi-steady state and does not have any effect on the results [7]. Time constants are in seconds, reactance are p.u. while all the other parameters are dimensionless.

TABLE 1Power system model parameters.

x_T	x_e	x_d	x_q	$x_{d}^{'}$	T'_{d0}	T'_{q0}	K_A	T_A	T_E	K_{f}	T_{f}	PF	D	k
0.15	0.34	1	1	0.18	5	1.5	30	0.4	0.56	0.1	1.3	0.95	0.05	0.1

⁹⁴⁷

Numerical continuation is applied to the nominal model using the parameter l as bifurcation parameter. The (non-zero) stable equilibrium point at l = 0 is found by simulating the model and this is provided as an initialization to COCO. The branch of equilibrium points as l is increased is reported in Fig. 5 by showing the values of three components of the state vector, namely E'_d , R_f , and V_L .



FIG. 5. Bifurcation diagram for the nominal power system model.

The analyses show that the system has a branch of stable equilibria for low values 953 of l (this part of the branch is denoted by a solid line), which undergoes a Hopf 954 bifurcation at $l_H=0.83$ (circle marker), with a frequency of the associated imaginary 955 eigenvalues equal to $\omega_H=2.6 \frac{rad}{s}$, and a saddle node bifurcation at $l_{SN}=1.13$ (square marker). As aforementioned, the model used in here is not exactly the same as that 956 957 of [13] as insufficient information was provided in that reference to reproduce their 958 results exactly. In [13], the Hopf bifurcation also occurred at a lower loading level 959 than the saddle node one but at different values, i.e. $l_H=0.37$ and $l_{SN}=1.03$. Thus, 960 qualitatively speaking, the results from Fig. 5 are similar to those in [13] (see also the 961 sensitivity analysis discussed next) and should enable the proposed robust bifurcation 962 margin approach to be tested by comparing with the results from $[13]^1$. 963

4.1.2. Sensitivity analysis of the Hopf bifurcation. The authors in [13] 964 compute the sensitivity of both bifurcations to the model's parameters (the focus will 965 966 be here only on the analyses for the Hopf one). This computation is performed by first defining what is termed the loading margin to instability at l_0 (a value of the 967 bifurcation parameter l corresponding to a stable equilibrium) as $M(l_0) = l_H - l_0$. 968 The first-order sensitivity M_c of M to a generic parameter c (here c represents any 969 model parameter, in the present case those in Table 1) is then computed as the partial 970 derivative of M with respect to c evaluated at l_0 , i.e. $M_c(l_0) = \frac{\partial M}{\partial c}\Big|_{l=l_0}$. Its computation is performed using normal vectors to the manifold of Hopf bifurcation points and 971 972 essentially consists of a sensitivity of the critical eigenvalue. An approximation to this sensitivity can be computed as $\tilde{M}_c = \frac{M(l_0, c+\epsilon) - M(l_0, c)}{\epsilon}$, where $M(l_0, c+\epsilon) = l_H^{c+\epsilon} - l_0$ 973 974 and $l_{H}^{c+\epsilon}$ is the value of l at which a Hopf bifurcation occurs when the parameter c is 975 increased to $c + \epsilon$. The quantity \tilde{M}_c is thus a finite difference approximation of M_c 976 and can be computed via numerical continuation. The results of such a sensitivity 977 analysis are reported in Table 2 for the parameters previously listed in Table 1.

 TABLE 2

 Sensitivity of the Hopf bifurcation to model parameters (continuation-based).

	x_T	x_e	x_d	x_q	$x_{d}^{'}$	T'_{d0}	T'_{q0}	K_A	T_A	T_E	K_f	T_f	PF	D	k
\tilde{M}_c	-0.96	-1.3476	-0.05	-0.006	-0.9111	0.039	0.0047	0.003	-0.2975	-0.1982	2.1	-0.14	1.5	-0.005	0.11

¹A MATLAB implementation of the power system model presented in this section, together with a file to run continuation analyses with COCO, is available at https://github.com/AndreaIan/PowerSystem_cont

It is noted that the sign of all the sensitivities (a negative sign means that an increase of the parameter makes the loading margin to instability decrease) coincide with those reported in [13] except for k, and the magnitude (proportional to the sensitivity to that parameter) is also generally well captured.

In order to show the connection between the sensitivity approach used in [13] and 982 the concept of robust bifurcation margin, a first type of analysis is discussed next. A 983 set of four parameters from the power system model is considered, namely x_q , K_A , 984 T_A , and K_f . Without loss of generality, only a subset of the parameters in Table 985 2 is selected to allow a more clear interpretation of the results. A subcritical value 986 of the loading level at which robustness of the plant is studied is then selected; this 987 988 is denoted l_0 according to the notation adopted in section 3.1. In all the analyses presented here the value $\bar{l}_0 = 0.725 < l_H$ will be considered. Once the set of uncertain 989 parameters and a value of the bifurcation parameter is selected, the corresponding 990 LFT can be constructed. It is observed that the dependence of the vector field on the 991 states cannot be captured directly in an LFT fashion. This is due to the trigonometric 992 functions (4.2-4.3a-4.3b) and square root (4.3c). For this reason, Taylor expansions 993 994 of these functions about the equilibrium state at l_0 are considered. The order of the expansion (1 and 2 depending on the specific state) is selected in order to guarantee 995 a satisfactory trade-off between accuracy and size of the LFT $\mathcal{F}(M_{\tilde{I}}, \Delta)$. For all 996 the uncertain parameters a range of variation of $\pm 15\%$ from the nominal value is 997 considered. 998

999 Program 3.2 is employed with an initialization provided by the nominal values of the equilibrium point and of the uncertainties. The value of the Lyapunov coefficient 1000 l_1 will not be considered as a variable in these analyses since the goal is not to study 1001 the effect of the parameters on the type of Hopf bifurcation, even though this would 1002 also be possible within this framework. Five different tests will be considered: four in 1003 which only one parameter belongs to the uncertainty set Δ_{μ} (the total size of each of 1004 1005 the four LFTs is 17), and one in which all the four parameters are included in Δ_u (the total size of the LFT is 25). The results are reported in Table 3 in terms of robust 1006 stability margin k_m , frequency $\hat{\omega}$ and worst-case perturbation for the normalized 1007 uncertainties. 1008

r		nad			-	-
test	k_m	$\hat{\omega} \frac{ruu}{s}$	δ_{x_q}	δ_{K_A}	δ_{T_A}	δ_{K_f}
1	24.3	2.3	24.3	•	•	•
2	∞	n.a.	•	n.a.	•	•
3	7.8	2.1	•	•	7.8	•
4	2.5	2.6	•	•	•	-2.5
5	1.54	2.2	1.54	-1.54	1.54	-1.54

TABLE 3 Sensitivity analysis with the robust bifurcation margin at $\bar{l}_0 = 0.725$.

The value of k_m for the first four tests, where only one parameter at a time 1009 1010 is allowed to vary, can be considered as a measure of the sensitivity of the Hopf bifurcation to that parameter – and it is thus expected to show similar results to 1011 1012 those obtained in [13]. Indeed, all the predictions reported in Table 2 (which was in agreement with [13]) are confirmed: high sensitivity to K_f , medium sensitivity to T_A , 1013 and practically no sensitivity to x_q and K_A (note that for the latter the optimization 1014 problem was found infeasible). Moreover, the signs of the worst-case perturbations 10151016 are also in agreement with the findings in Table 2. The fifth test shows that when 1017 all the parameters are acting together the margin k_m decreases, but it is still greater 1018 than 1, that is the power system at \bar{l}_0 is robust to the uncertainty considered. For the 1019 predicted worst-case perturbation, the Hopf bifurcation taking place at \bar{l}_0 is associated 1020 with a frequency $\hat{\omega} = 2.2 \frac{rad}{s}$ (recall that this is one of the optimization variables of 1021 Program 3.2), which is smaller than the one in the nominal case, but within the same 1022 frequency range.

While the proposed robust bifurcation margin framework can be used to retrieve 1023 the results of the sensitivity tests performed in [13], one of its advantage is that allows 1024 also for another type of sensitivity analysis. In particular, the effect of a parameter 1025on the bifurcation is evaluated while simultaneously accounting for the other uncer-1026 tainties affecting the system. This is inherently different from the sensitivity measure 1027 1028 proposed in [13], which is a first-order approximation of the partial derivative of the margin, and thus effectively neglects any coupling among the uncertainties. This key 1029 aspect will be exemplified with a second type of k_m -based analysis. 1030

It is known that the structured singular value μ can be used to evaluate the sensitivity of an instability to a set of n_{δ} selected parameters by performing multiple 1032 μ tests. This can be achieved for example using the skew- μ concept [30], or, within 1033 1034 standard μ analysis tools, by considering two different uncertainty levels w_{1,d_i} and w_{2,d_i} (recall the definition of the uncertainty level in Eq. 3.3) for each parameter d_i 1035 $(i = 1, ..., n_{\delta})$. In the first μ test (termed base to indicate it is the baseline test), all 1036the parameters have the uncertainty level w_{1,d_i} , while in the following n_{δ} tests, the 1037uncertainty level of the *i*-th parameter is set to w_{2,d_i} and for all the others it is kept 1038 1039 at w_{1,d_i} (with $j \neq i$). The difference between the peak of the baseline μ plot and the peaks of the other n_{δ} tests is proportional to the sensitivity of the instability to 1040the considered parameter. See [26] for an application of this analysis approach to the 1041 robust flutter problem. 1042

In the same spirit, the parameters studied in Table 3 are analyzed here considering $w_1 = 0.15$ (i.e the previously defined 15% uncertainty range) and $w_2 = 0.3$ (i.e. doubling the range for the specific parameter used in the n_{δ} test). Program 3.2 is again employed and the results are shown in Table 4 (the first column identifies the test performed, i.e. *base* and then the parameter whose uncertainty level is set to w_2).

test	; .	k_m	$\hat{\omega} \frac{rad}{s}$	δ_{x_q}	δ_{K_A}	δ_{T_A}	δ_{K_f}
bas	e 1	.54	2.2	1.54	-1.54	1.54	-1.54
x_q	1	.41	2.2	1.41	-1.41	1.41	-1.54
KA	1	.31	2	1.31	-1.31	1.31	-1.31
T_A	1	.24	2.2	1.24	-1.24	1.24	-1.24
K_f	().97	2.4	0.97	-0.97	0.97	-0.97

TABLE 4 Robust bifurcation margin sensitivity analysis at $\bar{l}_0 = 0.725$.

1048

The baseline test coincides with test 5 in Table 3 but is reported to facilitate the comparison. The different sensitivity of the considered parameters is confirmed in this new analysis (from the least sensitive parameter x_q to the most sensitive one K_f). However, it is also clear that every parameter now has an effect on the shift of the bifurcation point towards \bar{l}_0 . This is clearly seen comparing Table 3 and 4, where for the former table only K_f showed a high sensitivity effect (close to the value of the baseline test), but as shown in Table 4, when the uncertainty coupling is taken into account for the analysis, then all of the four parameters have similar levels to the

baseline case. This finding results from taking into account perturbations in the other 1057 1058 parameters while computing the parameter's sensitivity, and shows that the coupling among the uncertainties (not captured with first-order sensitivity approaches) can drastically affect the importance of some parameters. Specifically, parameters deemed 1060 unimportant with a first-order analysis can instead have a non-negligible impact on 1061 the bifurcation point. 1062

To further characterize this aspect, Figure 6 depicts the reciprocal of the robust 1063 bifurcation margin k_m as a function of the frequency. The five curves represent the 1064five cases considered in Table 4 and, unlike the *one-shot* tests discussed therein, are 1065obtained by fixing the frequency in the optimization and computing the value of the 1066 margin at each frequency.





FIG. 6. Sensitivity analysis of four parameters based on the robust bifurcation margin.

1067

The curves in Figure 6 resemble those typically employed in linear robust analysis 1068 with μ [48, 2, 26]. This points out once again the connection between the proposed 1069 concept of robust bifurcation margin k_m and the structured singular value μ . In particular, when $\frac{1}{k_m} \ge 1$, a perturbation in the allowed range of uncertainties exists 1070 1071 such that a Hopf bifurcation is experienced by the system when perturbed. Note that 1072 the peak of each curve coincides with the reciprocal of the margin reported in Table 1073 4. This representation allows the different sensitivities to the parameters discussed 1074previously to be immediately inferred. 1075

4.2. Aeroelastic system. 1076

4.2.1. Model description. The typical section is an aeroelastic case study com-1077 monly used for flutter analysis purposes [4], and consists of a rigid airfoil with lumped 1078 springs simulating the 3 structural degrees of freedom (DOFs): plunge h, pitch α and 1079trailing edge flap β . By defining the vector of structural states $x_s = [\frac{h}{h}; \alpha; \beta]$ and 1080 1081 aerodynamic states x_a (used to capture the unsteady aerodynamic contribution), the system can be described in matrix form as: 1082

1083 (4.4)
$$\dot{x} = \begin{bmatrix} \dot{x}_s \\ \ddot{x}_s \\ \dot{x}_a \end{bmatrix} = \begin{bmatrix} 0 & I & 0 \\ -M^{-1}K & -M^{-1}B & M^{-1}D \\ 0 & E & R \end{bmatrix} \begin{bmatrix} x_s \\ \dot{x}_s \\ x_a \end{bmatrix} = \mathcal{A}x,$$

-Bas

1

where M, B and K are respectively the aeroelastic inertial, damping and stiffness matrices:

$$M = M_s - \frac{1}{2}\rho_{\infty}b^2A_2,$$

1086 (4.5)
$$B = -\frac{1}{2}\rho_{\infty}bVA_1,$$

$$K = K_s - \frac{1}{2}\rho_{\infty}V^2A_0.$$

They include the structural mass M_s and stiffness matrices K_s plus the aerodynamic quasi-steady matrices A_i (ρ_{∞} is the air density and b the half chord distance). D, E, and R in (4.4) come from the rational approximation of the unsteady aerodynamic operator. The parameters defining the model are provided in [25] and the total state size n_x is 9 (6 structural and 3 aerodynamic). The interested reader is referred to [25] for a complete definition of the parameters defining the model and further details on aeroelastic modeling with uncertainties.

Nonlinearities in K_s are considered in this work. Specifically, hardening cubic terms for the plunge and pitch degrees of freedom are assumed, and the matrix K_s is rewritten accordingly:

1097 (4.6)
$$K_s = K_s^L + K_s^{NL} = \begin{bmatrix} K_h^L & 0 & 0 \\ 0 & K_\alpha^L & 0 \\ 0 & 0 & K_\beta \end{bmatrix} + \begin{bmatrix} K_h^{NL} K_h^L \left(\frac{h}{b}\right)^2 & 0 & 0 \\ 0 & K_\alpha^{NL} K_\alpha^L \alpha^2 & 0 \\ 0 & 0 & 0 \end{bmatrix},$$

where K_h , K_α and K_β are respectively the plunge, pitch and control surface stiffness. As is common practice [11], the coefficients of the nonlinear terms are assumed proportional to the corresponding linear ones through the dimensionless coefficients K_h^{NL} and K_α^{NL} (assumed here equal to 100). The hardening effect modelled in (4.6) takes into account the fact that the stiffness properties change when the system undergoes large deformations, with an increase in the stiffness generally observed.

The dynamics of the system is thus in the form of the generic vector field (2.1), and, by selecting the speed V as bifurcation parameter, it holds:

1106 (4.7)
$$\dot{x} = f(x, V) = \mathcal{A}^{L}(V)x + f^{NL}(x),$$
$$J(x, V) = \mathcal{A}^{L}(V) + \nabla_{x} f^{NL}(x),$$

1107 where: $\mathcal{A}^L : \mathbb{R} \to \mathbb{R}^{n_x \times n_x}$ is obtained from \mathcal{A} (4.4) by setting the nonlinear terms 1108 to zero; $f^{NL} : \mathbb{R}^{n_x} \to \mathbb{R}^{n_x}$ is the nonlinear part of the vector field; and the state is 1109 $x = [x_s; \dot{x}_s; x_a]$. Note that, for the nonlinearities considered here (4.6), f^{NL} (and thus 1100 also $\nabla_x f^{NL}$) does not depend on the speed.

Following the notation in section 3.1, V_H will denote the speed at which the 1111 nominal system undergoes a Hopf bifurcation, and after which it will potentially 1112exhibit limit cycle oscillations. Given a subcritical speed \bar{V}_0 (such that $\bar{V}_0 < V_H$ 1113corresponds to a stable equilibrium) and the definition of a vector δ of parametric 1114 uncertainties, then the distance in the parameter space of the equilibrium at V_0 from 1115 the closest Hopf bifurcation is computed by means of the robust bifurcation margin 11161117 k_m . The robust bifurcation analysis will thus allow the quantification of the influence of parametric uncertainties on the onset of LCO, which are a notorious problem for 1118 nonlinear aeroelastic systems [11]. 1119

Numerical continuation can be applied to (4.7) after having specified the value of the trim state x_t . Two cases will be considered, case 1 (c1) with $x_t = 0$ and case

2 (c2) featuring a non-zero value $\alpha_t = 1^\circ$ for the angle of attack of the section. The 1122latter is physically motivated by the fact that the section is generating positive lift to 11231124 counterbalance gravitational forces directed downwards. Figure 7 shows the standard (i.e. nominal) bifurcation diagrams with V on the x-axis and the normalized plunge 1125DOF $\frac{h}{h}$ on the y-axis (in the case of branches of LCO solution branches, this is the 1126 maximum value over a period). The usual convention of representing stable steady-1127 states (equilibria and LCOs) with solid lines and unstable ones with dashed lines is 1128 adopted, and the Hopf bifurcation is marked with a circle. 1129



FIG. 7. Bifurcation diagram for the nominal vector field for two different trim states.

1130 The system experiences supercritical Hopf bifurcations at

1131 $V_{H}=302.7 \frac{m}{s}$ for c1 and $V_{H}=289.0 \frac{m}{s}$ for c2. The frequency of the associated 1132 imaginary eigenvalues are respectively $\omega_{H}=70 \frac{rad}{s}$ and $\omega_{H}=75 \frac{rad}{s}$.

4.2.2. Computation of robust bifurcation margins. Including uncertainties
 in the nominal vector field of (4.7) yields the expression for the uncertain vector field

1136 (4.8a) $\dot{x} = \tilde{f}(x, V, \delta) = \tilde{\mathcal{A}}^L(V, \delta) x + \tilde{f}^{NL}(x, V, \delta),$

1137 (4.8b)
$$\tilde{J}(x,V,\delta) = \tilde{\mathcal{A}}^L(V,\delta) + \nabla_x \tilde{f}^{NL}(x,V,\delta).$$

1139 The bifurcation parameter V will be fixed in the subsequent analyses to $\bar{V}_0 = 270 \frac{m}{s}$, 1140 which, recall Figure 7, is associated in both cases with stable equilibria –and hence, 1141 it is a valid choice according to the discussion in section 3.1.

1142 The initial step to compute robust bifurcation margins is the definition of the 1143 nominal system and of the uncertainty set, which in turn will drive the construction of 1144 the underlying LFT. The former is described by (4.7), while the uncertainty definition 1145 is chosen to define a range of variation of $\pm 10\%$ from the nominal value for the 1146 coefficients $M_{s_{11}}$, $M_{s_{22}}$, $K_{s_{22}}$ and of $\pm 5\%$ for $M_{s_{12}}$ and $K_{s_{11}}$

1147 (4.9)
$$\Delta_u = \operatorname{diag}(\delta_{K_{s_{10}}^L}, \delta_{M_{s_{11}}}, \delta_{M_{s_{12}}}, \delta_{M_{s_{22}}}).$$

1148 This uncertainty definition was considered since it is the same as that used in [25], 1149 where the linear problem (i.e. $K_h^{NL} = K_\alpha^{NL} = 0$) was extensively analyzed by means ¹¹⁵⁰ of nominal (eigenvalue analysis) and robust (μ analysis) techniques. This testcase is ¹¹⁵¹ thus used to benchmark the first set of numerical results obtained with the method

1152 proposed in this paper.

Program 3.2 is computed with an initialization provided by the nominal values of the equilibrium point and of the uncertainties. The results from the program are reported in Table 5 in terms of robust stability margin k_m , frequency $\hat{\omega}$ of the imaginary eigenvalues at \bar{V}_0 , and type of Hopf bifurcation. Recall indeed that Program 3.2 calculates the closest Hopf bifurcation without constraining the value of l_1 , and the type of predicted Hopf bifurcation was assessed a posteriori with numerical continuation of the perturbed system.

TABLE 5 Robust bifurcation margins at $\bar{V}_0 = 270 \frac{m}{s}$ for uncertainties in the set (4.9).

	k_m	$\hat{\omega} \frac{rad}{s}$	type
c1	0.73	71.5	super
c2	0.49	75.1	super

1159

1160 It is inferred from the first column of Table 5 that in both cases the Hopf bifur-1161 cation could be shifted to $V = 270 \frac{m}{s}$ within the uncertainty range (note indeed that 1162 $k_m < 1$). Another important observation is that c1 gives a margin k_m within less 1163 than 1% of the result from the literature [25] (obtained with μ considering the linear 1164 system at the same speed $V = 270 \frac{m}{s}$). The (normalized) uncertainty vector found 1165 here by the optimizer is

1166 (4.10)
$$\hat{\delta} = [\delta_{K_{s_{22}}^L}; \delta_{K_{s_{11}}^L}; \delta_{M_{s_{11}}}; \delta_{M_{s_{12}}}; \delta_{M_{s_{22}}}], \\ = [-0.7328; 0.7328; -0.7328; 0.5027; 0.7328],$$

which also features the same perturbations (within a small tolerance) as those de-1167 1168 tected in [25] (their physical meaning in relation to the onset of flutter was discussed in the reference). In order to better appreciate the importance of this result, let 1169 us recall that nominal analyses (Figure 7) found for c1 the branch of equilibria at 1170 x = 0 regardless of V. Since the uncertainties selected here only affect $\tilde{\mathcal{A}}^L$, then $\tilde{f}^{NL}(0,V,\cdot) = f^{NL}(0,V) = 0$ and thus $\nabla_x \tilde{f}^{NL} \equiv 0$. That is, the determination of 11711172 k_m is equivalent (for this specific case) to the problem solved by μ , i.e., finding the 1173smallest perturbation matrix such that $\tilde{\mathcal{A}}^L$ is neutrally stable. The good matching 1174 with the literature results is very important, since in [25] μ_{LB} and μ_{UB} were shown 1175to be close, indicating that the true value of μ was determined. This result hence ver-1176 ifies the correctness of the approach proposed here since it recovers the result which, 1177for this specific case, is known a priori to be the correct one. Moreover, at least for 1178 this case, Program 3.2 is able to detect the global minimum of the optimization. 1179

1180 Another positive feature is that Program 3.2 has the frequency ω as a decision vari-1181 able, whereas μ was applied in [25] at discrete frequencies because this is the available 1182 implementation for the standard algorithms [2] (which has the drawback of possibly 1183 missing critical frequencies and thus overestimating the value of the stability margin).

1184 Case c2 is then considered (with $\alpha_t = 1^\circ$). This cannot be analyzed with μ 1185 because \tilde{J} is now also a function of the nonlinear terms due to non zero values for 1186 the equilibria (which in turn depend on the uncertainty). For this reason, it is not 1187 possible to compare the results with the true analytic solution. However, it is noted 1188 that k_m now achieves a smaller value than for c1. This is in accordance with the 1189 nominal analyses in Figure 7, for which c2 presented a smaller V_H than c1. Thus,

30

1190 as $\overline{V} = 270 \frac{m}{s}$ is closer to the nominal bifurcation speed for c2, a smaller robustness 1191 margin is expected. Note also that the two predicted frequencies $\hat{\omega}$ are relatively close 1192 to the nominal ones. These interpretations thus give some confidence that an accurate 1193 prediction of the margin has also been obtained for c2.

Other important information gathered in Table 5 is the type of closest Hopf bifurcations. Note that in order to obtain this result, the solver COCO was used to perform numerical continuation of the perturbed system, which also allowed verification that the latter experienced a Hopf bifurcation at $\bar{V}_0 = 270 \frac{m}{s}$, as expected.

These analyses show that the closest Hopf bifurcations are of the same nature as the corresponding ones in nominal conditions. Based on the greater attention typically devoted to subcritical LCOs due to the associated risks [11], the following analyses will make use of Program 3.4 to investigate whether changes in parameter values can drive the Hopf bifurcation from supercritical to subcritical. Without loss of generality, only the case c^2 will be considered.

1204 Uncertainties in two aerodynamic parameters are added to the set (4.9), namely, 1205 the terms $A_{0_{12}}$ and $A_{0_{22}}$ of the steady aerodynamics matrix A_0 (4.5). These corre-1206 spond to the lift and moment coefficients of the airfoil respectively, and are allowed 1207 to vary within $\pm 20\%$ from their nominal values. Table 6 shows the solutions provided 1208 by Program 3.4 for the two types of possible Hopf bifurcation in terms of: Lyapunov 1209 coefficient l_1 , stability margin k_m , frequency $\hat{\omega}$, and normalized perturbations. A 1209 to the value of the Lyapunov coefficient was used.

 $\begin{array}{c} {\rm TABLE} \ 6 \\ Worst-case \ perturbations \ and \ margins \ to \ supercritical \ and \ subcritical \ Hopf \ bifurcations. \end{array}$

	l_1	k_m	$\hat{\omega} \frac{rad}{s}$	$\delta_{K^L_{s_{22}}}$	$\delta_{K^L_{s_{11}}}$	$\delta_{M_{s_{11}}}$	$\delta_{M_{s_{12}}}$	$\delta_{M_{s_{22}}}$	$\delta_{A_{0_{22}}}$	$\delta_{A_{0_{12}}}$
super	-10^{3}	0.25	76	-0.25	0.25	-0.25	-0.25	0.25	0.25	-0.25
sub	1	3.13	67	-3.12	3.12	-3.12	-3.12	3.12	3.12	1.83

1210

The supercritical case is consistent with the corresponding case in Table 5. Indeed, 1211 the margin approximatively halves as a result of the additional uncertainty in the 1212 system, while the frequency has a similar value. Note also that the constraint on l_1 1213 is not active and thus l_1 has a large absolute value. On the contrary, the subcritical 1214 case features a far higher margin (which, according to the definition of k_m given in 1215subsection 3.1, points out that there is no perturbation inside the allowed set capable 1216 of prompting the investigated bifurcation) and achieves a value of l_1 equal to the 1217 tolerance ϵ_l . Another interesting fact is that while all the normalized perturbations 1218 feature the same sign as in the supercritical case, this does not hold for $A_{0_{12}}$ which 1219 1220 has an opposite perturbation and, in absolute value, smaller than the others. This is an interesting aspect, because according to standard interpretations of unstable 1221 aeroelastic phenomena [4, 25], a negative perturbation for $A_{0_{12}}$ would be expected 1222 (as noted for the supercritical case). The justification for this could be sought in 1223 the physical mechanisms prompting subcritical LCO [11] and will be investigated in 1224 1225future studies. It is remarked here that the commented scenario is distinctive of this problem, where different (possibly conflicting) constraints define the worst-case 12261227conditions. While robustness in the linear context focuses on the loss of stability only, from a dynamical systems perspective this becomes a multi-faceted concept 1228 characterized by concurrent conditions and thus non-intuitive results can be found. 1229

Figure 8 shows bifurcation diagrams relative to worst-case combinations of parameters found by Program 3.4 by changing the tolerance on the Lyapunov coefficient



1232 ϵ_l . In the legend of Figure 8, the value of the Lyapunov coefficient at the bifurcation point is indicated.

FIG. 8. Bifurcation diagram of the system for different worst-case perturbations.

1233 1234

The first important observation is that all the cases display a Hopf bifurcation 1234 at $\bar{V}_0 = 270 \frac{m}{s}$. The branches relative to the solutions from Table 6 (obtained with $\epsilon_l = 1$) are $l_1 = -10^3$ and $l_1 = 1$. This in turn demonstrates that Program 3.4 is able 1235 1236 to correctly predict worst-case combinations of uncertainty which lead to respectively 1237 supercritical and subcritical bifurcation. For the other curves $l_1 = \epsilon_l$ holds since this 1238 constraint is always active, and the associated margins k_m slightly increase compared 1239 to the value 3.13 featured in Table 6. It is stressed that a quantitative interpreta-1240 tion of the absolute value of l_1 depends on the arbitrary normalization adopted for 12411242 the eigenvector q in its definition (3.10). The point made here is qualitative and, specifically, is that as the tolerance ϵ_l (and thus l_1) is increased, the subcritical Hopf 1243 bifurcation predicted by the optimizer is more pronounced (i.e. the range of speeds 1244 for which unstable and stable LCOs coexist with the branch of stable equilibria is 1245larger). Even though this is not guaranteed by the Hopf bifurcation theorem, since l_1 1246 is defined on the center manifold at the bifurcation point only, the magnitude of the 1247Lyapunov coefficient can be taken as a measure of the subcriticality of the LCO (when 1248 comparing different instances computed with the same normalization of q). Figure 1249 8 shows therefore that embedding the constraint on the Lyapunov coefficient in the 1250bifurcation margin computation is successfully done by the optimization. 1251

The last part of the section is aimed at providing insights into the numerical aspects of the algorithms. As a preamble, it is observed that there are not definitive answers with respect to robustness to local minima or efficiency of the algorithms as these will depend on many aspects such as, for example, the type of vector field (not only size and degree, but also number of attractors) and the optimization algorithm employed (which is an aspect that has not been investigated in this work). Investigation of these important features are left for future work.

The execution time of Program 3.4 is larger than that of Program 3.2 (approximatively 6s against 3s for the case with 7 uncertainties). Most importantly, the addition of the constraint on l_1 exacerbates the issue of local minima, especially when this is an active constraint. The set of strategies described in Section 3.3 were thus employed to obtain the results presented in Figure 8. Specifically, reinitializing the optimization with points on the auxiliary manifold \mathcal{H}_g and with solutions obtained by fixing the frequency in Program 3.4 led to significant improvements in the solution.

Finally, it is remarked that for all the analyzed cases, the worst-case combinations of the uncertain parameters predicted by the optimization problem were used to perform numerical continuation analyses of the perturbed system with COCO. In all cases the perturbed systems encountered a Hopf bifurcation at the pre-selected speed \bar{V}_0 . Even though this fact does not ensure that the global optimum (i.e. smallest margin to bifurcation) was found, it represents important evidence of the validity of the overall approach.

5. Conclusions. The paper develops a framework for the analysis of nonlinear 1273 systems subject to parametric uncertainties with the goal of studying robustness of 1274stable equilibria to the onset of dynamic bifurcations. A scalar metric quantifying a 12751276 perturbation in the uncertainty set is first defined, and the magnitude of the smallest perturbation such that a stable equilibrium is driven into a Hopf bifurcation point is 1277 named the robust bifurcation margin k_m . Its definition, which also allows the nature 1278of the closest Hopf bifurcation (subcritical or supercritical) to be specified, is based 1279on the idea of building a Linear Fractional Transformation model of the uncertain 1280 1281 Jacobian and studying its singularity. The proposed margin can be interpreted as an extension of the structured singular value μ to the nonlinear context. The compu-1282 tation of k_m is recast as a nonlinear smooth constrained optimization problem, and 1283 as such it suffers in principle from the issue of local minima. Thus, the proposed 1284programs technically provide only an upper bound on the margin. However, several 12851286 mitigation strategies are described in order to tighten the gap with the actual margin, 1287 including a continuation-based multi-start strategy. Application of the framework is demonstrated on two case studies: a power system model and an aeroelastic system 1288 exhibiting nonlinear flutter behaviour. For the former, analyses show that k_m can be 1289used to infer sensitivity of the Hopf bifurcation to system's parameters and it allows 1290 1291more accurate predictions than those achieved with available methods only providing 1292first-order information. As for the latter, first the same results obtained in the literature with μ are retrieved, and then the possibility to distinguish between closest 1293 1294 subcritical and supercritical bifurcations is explored. The results verify from different perspectives soundness of the newly introduced concept and provide examples of its 1295 perspective advantages over available techniques to study the nonlinear robustness 1296 1297 problem in different application domains.

1298

REFERENCES

- [1] G. B. ARFKEN AND H. J. WEBER, Mathematical methods for physicists, Academic Press; 4th
 ed., San Diego, CA, 1995.
- 1301 [2] G. BALAS, R. CHIANG, A. PACKARD, AND M. SAFONOV, Robust Control Toolbox, 2009.
- [3] W. BEYN, A. CHAMPNEYS, E. DOEDEL, W. GOVAERTS, Y. A. KUZNETSOV, AND B. SANDSTEDE, Numerical continuation, and computation of normal forms., in Handbook of Dynamical Systems, Vol 2 / B. Fiedler (edit.), Elsevier, 2002, Chapter 4. - ISBN 0-444-50168-1, 2002, pp. 149–219.
- 1306 [4] R. L. BISPLINGHOFF AND H. ASHLEY, Principles of Aeroelasticity, Wiley, 1962.
- [5] R. BRAATZ, P. YOUNG, J. DOYLE, AND M. MORARI, Computational-complexity of μ-calculation,
 IEEE Transactions on Automatic Control, 39 (1994), pp. 1000–1002.
- [6] C. A. CANIZARES, Calculating optimal system parameters to maximize the distance to saddlenode bifurcations, IEEE Transactions on Circuits and Systems, 45 (1998), pp. 225–237.
- 1311[7] J. H. CHOW AND A. GEBRESELASSIE, Dynamic voltage stability analysis of a single machine1312constant power load system, in 29th IEEE Conference on Decision and Control, 1990.
- 1313 [8] G. CIRILLO, G. HABIB, G. KERSCHEN, AND R. SEPULCHRE, Analysis and design of nonlinear

- resonances via singularity theory, Journal of Sound and Vibration, 392 (2017), pp. 295–306.
 H. DANKOWICZ AND F. SCHILDER, An extended continuation problem for bifurcation analysis in the presence of constraints, ASME 2009 International Design Engineering Technical
- Conferences and Computers and Information in Engineering Conference, 2009.
 H. DANKOWICZ AND F. SCHILDER, *Recipes for Continuation*, Society for Industrial and Applied Mathematics, Philadelphia, PA, 2013.
- 1320 [11] G. DIMITRIADIS, Introduction to Nonlinear Aeroelasticity, Aerospace Series, Wiley, 2017.
- [12] I. DOBSON, Computing a closest bifurcation instability in multidimensional parameter space,
 Journal of Nonlinear Science, 3 (1993), pp. 307–327.
- 1323[13] I. DOBSON, F. ALVARADO, AND C. L. DEMARCO, Sensitivity of hopf bifurcations to power1324system parameters, in [1992] Proceedings of the 31st IEEE Conference on Decision and1325Control, 1992.
- [14] E. J. DOEDEL, T. F. FAIRGRIEVE, B. SANDSTEDE, A. R. CHAMPNEYS, Y. A. KUZNETSOV,
 AND X. WANG, Auto-07p: Continuation and bifurcation software for ordinary differential equations, tech. report, 2007.
- [15] J. DOYLE, Analysis of feedback systems with structured uncertainties, IEE Proceedings D Con trol Theory and Applications, 129 (1982), pp. 242–250.
- [16] J. GERHARD, W. MARQUARDT, AND M. MONNIGMANN, Normal vectors on critical manifolds for robust design of transient processes in the presence of fast disturbances, SIAM Journal on Applied Dynamical Systems, 7 (2008), pp. 461–490.
- 1334 [17] P. GILL, W. MURRAY, AND M. WRIGHT, Practical optimization, Academic Press, 1981.
- [18] M. GOLUBITSKY AND D. SCHAEFFER, Singularities and Groups in Bifurcation Theory, Applied
 Mathematical Sciences, Springer-Verlag New York, 1985.
- [19] W. GOVAERTS, Numerical Methods for Bifurcations of Dynamical Equilibria, Society for In dustrial and Applied Mathematics, 2000.
- R. GRAY, A. FRANCI, V. SRIVASTAVA, AND N. E. LEONARD, Multiagent decision-making dynamics inspired by honeybees, IEEE Transactions on Control of Network Systems, 5 (2018), pp. 793–806.
- [21] J. GUCKENHEIMER AND P. HOLMES, Nonlinear Oscillations, Dynamical Systems, and Bifurcations of Vector Fields, Applied Mathematical Sciences, Springer New York, 2002.
- 1344[22] M. HAYES, D. BATES, AND I. POSTLETHWAITE, New tools for computing tight bounds on the1345real structured singular value, Journal of Guidance, Control, and Dynamics, 24 (2001),1346pp. 1204–1213.
- 1347 [23] R. HORN AND C. JOHNSON, Matrix Analysis, Cambridge University Press, 1990.
- [24] A. IANNELLI, M. LOWENBERG, AND A. MARCOS, An extension of the structured singular value to nonlinear systems with application to robust flutter analysis, 5th CEAS Conference on Guidance, Navigation and Control (EuroGNC), 2019.
- [25] A. IANNELLI, A. MARCOS, AND M. LOWENBERG, Aeroelastic modeling and stability analysis:
 1352 A robust approach to the flutter problem, International Journal of Robust and Nonlinear
 1353 Control, 28 (2018), pp. 342–364.
- [26] A. IANNELLI, A. MARCOS, AND M. LOWENBERG, Study of Flexible Aircraft Body Freedom Flutter
 with Robustness Tools, J. of Guidance, Control and Dynamics, 41 (2018), pp. 1083–1094.
- 1356 [27] Y. KUZNETSOV, Elements of Applied Bifurcation Theory, Springer-Verlag New York, 2004.
- [28] J. MAGNI, Linear fractional representation toolbox modelling, order reduction, gain scheduling,
 Technical Report TR 6/08162, DCSD, ONERA, Systems Control and Flight Dynamics
 Department, 2004.
- [29] A. MARCOS, D. BATES, AND I. POSTLETHWAITE, Nonlinear symbolic LFT tools for modeling, analysis and design, in Nonlinear Analysis and Synthesis Techniques in Aircraft Control, L. N. in Control and S.-V. Information Science, eds., 2007.
- 1363[30] A. MARCOS, D. BATES, AND I. POSTLEWHITE, Control oriented uncertainty modeling using1364 μ sensitivities and skewed μ analysis tool, in IEEE Conference on Decision and Control,13652005.
- 1366 [31] MATLAB, Optimization Toolbox User's Guide, 2014.
- [32] A. MAZZOLENI AND I. DOBSON, Closest bifurcation analysis and robust stability design of flexible
 satellites, Journal of Guidance, Control, and Dynamics, 18 (1995), pp. 333–339.
- [33] Z. MICHALEWICZ AND D. FOGEL, How ro Solve It: Modern Heuristics, Springer-Verlag, 2nd ed.,
 2004.
- [34] M. MONNIGMANN AND W. MARQUARDT, Steady-state process optimization with guaranteed robust stability and feasibility, AIChE Journal, 49, pp. 3110–3126.
- [35] A. PACKARD AND J. DOYLE, The Complex Structured Singular Value, Automatica, 29 (1993),
 pp. pp. 71–109.
- 1375 [36] A. PACKARD, M. FAN, AND J. DOYLE, A power method for the structured singular value, Proc.

- 1376 of the Conference on Decision and Control, December 1988.
- [37] A. ROBERTS, Computer algebra derives correct initial conditions for low-dimensional dynamical models, Computer Physics Communications, 126 (2000), pp. 187 – 206.
- [38] A. ROBERTS, Normal form transforms separate slow and fast modes in stochastic dynamical systems, Physica A: Statistical Mechanics and its Applications, 387 (2008), pp. 12 38.
- [39] C. Roos, Systems modeling, analysis and control (SMAC) toolbox: An insight into the robustness analysis library, in 2013 IEEE Conference on Computer Aided Control System Design (CACSD), 2013.
- [40] C. ROOS AND J. BIANNIC, A detailed comparative analysis of all practical algorithms to compute
 lower bounds on the structured singular value, Control Engineering Practice, 44 (2015),
 pp. 219–230.
- [41] W. D. ROSEHART AND C. A. CANIZARES, Bifurcation analysis of various power system models, International Journal of Electrical Power & Energy Systems, 21 (1999), pp. 171 – 182.
- [42] M. SAFONOV, Origins of robust control: Early history and future speculations, 7th IFAC Sym posium on Robust Control Design, June 2012.
- [43] M. SAFONOV AND M. ATHANS, A multiloop generalization of the circle criterion for stability
 margin analysis, IEEE Transactions on Automatic Control, 26 (1981), pp. 415–422.
- [44] P. SEILER, A. PACKARD, AND G. J. BALAS, A gain-based lower bound algorithm for real and mixed μ problems, Automatica, 46 (2010), pp. 493–500.
- 1395 [45] A. TAYLOR AND W. MANN, Advanced Calculus, Wiley, 1983.
- [46] V. VENKATASUBRAMANIAN, H. SCHATTLER, AND J. ZABORSZKY, Voltage dynamics: study of a generator with voltage control, transmission, and matched MW load, IEEE Transactions on Automatic Control, 37 (1992), pp. 1717–1733.
- [47] A. YAZICI, A. KARAMANCIOĞLU, AND R. KASIMBEYLI, A nonlinear programming technique to compute a tight lower bound for the real structured singular value, Optimization and Engineering, 12 (2011), pp. 445–458.
- 1402 [48] K. ZHOU, J. C. DOYLE, AND K. GLOVER, Robust and Optimal Control, Prentice-Hall, 1996.